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## Alloy design for laser powder bed fusion additive manufacturing: a critical review

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# Alloy design for laser powder bed fusion additive manufacturing: a critical review

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## 1. Abstract

Metal additive manufacturing (AM) has been extensively studied in recent decades. Despite the significant progress achieved in manufacturing complex shapes and structures, challenges such as severe cracking when using existing alloys for laser powder bed fusion (L-PBF) AM persisted. These challenges arise because commercial alloys are primarily designed for conventional casting or forging processes, overlooking the fast cooling rates, steep temperature gradients, and multiple thermal cycles of L-PBF. To address this, there is an urgent need to develop novel alloys specifically tailored for L-PBF technologies. This review provides a comprehensive summary of the strategies employed in alloy design for L-PBF. It aims to guide future research on designing novel alloys dedicated to L-PBF instead of adapting the existing alloys for L-PBF. The review begins by discussing the features of L-PBF processes, focusing on rapid solidification and intrinsic heat treatment. Next, the printability of the four main existing alloys (Fe-, Ni-, Al-, and Ti-based alloys) is critically assessed, with a comparison to their conventional weldability. It was found that the weldability criteria are not always applicable in estimating printability. Furthermore, the review presents recent advances in alloy development and associated strategies, categorizing them into crack mitigation-oriented, microstructure manipulation-oriented, and machine learning-assisted approaches. Lastly, an outlook and suggestions are given to highlight the issues that need to be addressed in future work.

**Keywords:** laser powder bed fusion, alloy design, printability, crack mitigation

## 2. Introduction

Additive Manufacturing (AM), building objects layer by layer directly from three-dimensional data [1], has been regarded as a revolutionary manufacturing technology since it was born in the 1980s [2]. The early studies primarily focused on polymers and AM was used for rapid prototyping applications. Later on, with the introduction of selective laser melting (SLM, also termed as laser powder bed fusion, L-PBF) in the 1990s [3], the AM technology has been widely used for the production of complex metal parts. Nowadays, the AM technology gained recognition for end-use part production, particularly in aerospace, automotive, and medical industries [4-6].

In contrast to traditional 'subtractive' manufacturing, AM offers notable advantages. AM allows for complex geometries, including internal cavities, undercuts, and intricate lattice structures, without the

constraints of traditional manufacturing methods. AM also enables quick iterations in the design process, reducing time and costs associated with tooling. Additionally, parts can be designed with specific material properties and optimized for their intended function, leading to lightweight and efficient designs [7-10].

Despite notable achievements in manufacturing complex shapes and structures using AM [11-14], the availability of printable metals that are crack-free in the as-built state is still limited. The presence of defects such as cracks and pores significantly compromises the mechanical properties, as the alloys were initially designed for wrought or casting without considering the unique characteristics of the AM process. Laser powder bed fusion (L-PBF) additive manufacturing involves high cooling rates, steep temperature gradients, and numerous thermal cycles [15, 16]. The compositions of many existing alloys are not well-suited for these characteristics, leading to cracking when directly applied to L-PBF. For instance, while boron is added to nickel-based superalloys in casting to enhance grain boundary cohesion and improve creep properties [17], its addition in L-PBF is detrimental, causing hot cracking due to reduced solidus temperature and the formation of a stabilized liquid film at lower temperatures [18]. Therefore, a novel alloy design approach is necessary, taking into account the processability during L-PBF.

Metal additive manufacturing has been extensively reviewed by various research groups from different perspectives, for instance the microstructure [19, 20], processing [21, 22], numerical modelling [23-25], mechanical properties [21, 26, 27], and post-treatments [28-30]. The alloy development has also been reviewed, including Ti-based [31-35], Al-based [20, 36], Ni-based [37, 38], Fe-based [2, 39], and Mg-based alloys [40, 41]. However, most of the work focuses on adapting existing alloys to the L-PBF process, rather than developing specific alloys dedicated to L-PBF. Furthermore, there is a lack of analysis on the correlation between elements and the printability of an alloy system in existing reviews. Additionally, less attention has been paid to comparing the weldability and printability of the existing alloys. Therefore, in this work, we first analyze the differences between conventional casting/welding processes and L-PBF process. Subsequently, the weldability and printability of commonly used alloy systems are analyzed, i.e., Fe-based, Ni-based, Al-based, and Ti-based, assessing the validity of using convention weldability to predict printability and identifying problems when using existing alloys in L-PBF. Subsequently, strategies used in alloy design for L-PBF are summarized and categorized into crack mitigation-oriented, microstructure manipulation-oriented, and machine learning-assisted approaches. At last, the current development of alloy design for L-PBF is summarized, and suggestions for future work are provided. The purpose of this review work is to provide guidance and new insights for designing novel alloys specifically

1 tailored for L-PBF.

### 3. Laser powder bed fusion

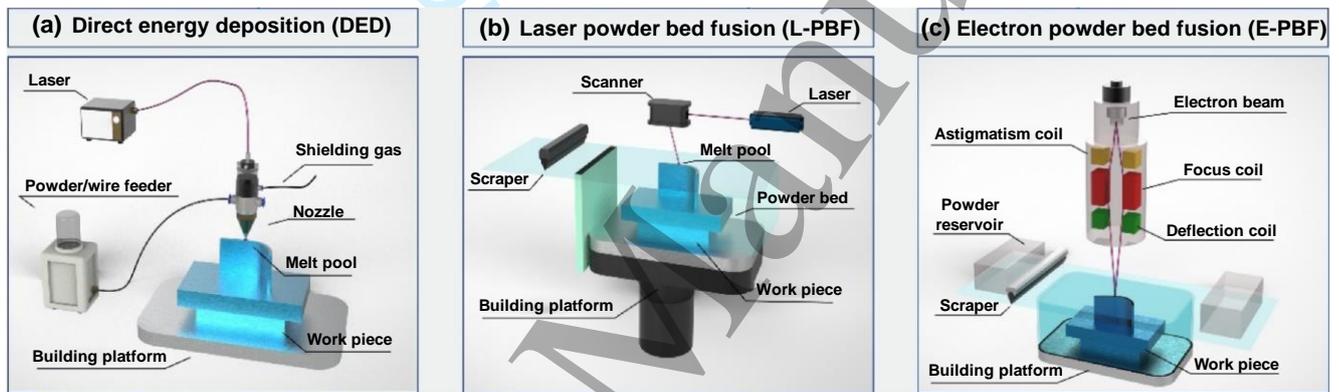
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6 The prevalent metal additive manufacturing technologies primarily include powder bed fusion (PBF)  
7 and directed energy deposition (DED), as shown in **Figure 1**. The metal powder in PBF is spread uniformly  
8 over a build platform, and each layer is selectively melted according to the 3D model. In DED, both metal  
9 powders and wires can be used to precisely deposit on a substrate or previously deposited layers through a  
10 nozzle or feedstock mechanism. In general, PBF processes typically offer higher resolution and accuracy,  
11 resulting in parts with fine details and intricate geometries, comparing to DED. However, DED  
12 technologies offer greater deposition freedom than PBF, as they can be used to fabricate and repair parts in  
13 a more flexible manner, enabling large-scale production.  
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23 The PBF can be further categorized into laser powder bed fusion (L-PBF) and electron powder bed fusion  
24 (E-PBF) based on the types of heat sources employed, as shown in **Figure 1**. The laser beam used in L-  
25 PBF is a coherent beam of light with specific wavelengths, typically in the infrared range. The electron  
26 beam used in E-PBF consists of accelerated electrons. It has a larger interaction volume and can penetrate  
27 deeper into the material, thus creating larger melt pools than laser. Furthermore, the build chamber during  
28 E-PBF can be heated to a much higher temperature, e.g., 1 000 °C, while most of the commercial L-PBF  
29 equipment can only preheat the baseplate to 200 °C. L-PBF typically exhibits high cooling rates due to the  
30 localized melting and rapid solidification of the metal powder. The localized melting and solidification  
31 result in steep thermal gradients between the molten and solidified regions. Therefore, L-PBF can produce  
32 fine-grained microstructures. In this review, we primarily focus on alloy design for the L-PBF technology,  
33 but will also cite multiple works on E-PBF or DED as a comparison or reference to inspire future alloy  
34 design work for L-PBF.  
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48 As shown in **Figure 1**, during L-PBF, a thin layer of metal powder, typically around 20 ~ 100  
49 micrometers thick, is evenly spread across a metal plate. A laser beam is precisely guided across the powder  
50 bed in a pattern defined by the 3D model. As the laser beam scans the powder bed, metallic powders absorb  
51 laser light and convert it into heat energy. It should be noted that part of the laser energy can reflect off the  
52 powder's surface due to its high reflectivity. When the temperature reaches or exceeds the powder's melting  
53 point, the particles start to melt, forming a localized molten pool and then cooling down rapidly. After each  
54 layer is melted and solidified, the build platform is lowered by a layer thickness. A new layer of metal  
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powder is then spread across the previously printed layer. The process is repeated, building the object layer by layer until it is fully formed. Metal powders used in L-PBF should have a controlled particle size and shape, as well as good flowability and homogeneity. These properties are crucial for uniform powder spreading and influence the laser-powder interaction and energy absorption, impacting the final microstructure and properties of the printed object [42].

Comparing to conventional casting, metallic parts undergo much higher cooling rates ( $10^3 \sim 10^8 \text{ }^\circ\text{C}\cdot\text{s}^{-1}$ )[43] and heating rates during L-PBF. Due to the layer-wise printing nature, the solidified section experiences multiple thermal cycles, which is termed as intrinsic heat treatment (IHT). The IHT generates substantial internal thermal stresses and dislocations within the solidified part, playing a significant role in the part properties. This section provides an overview of the key characteristics of L-PBF, highlighting the associated advantages and challenges.



**Figure 1.** Three widely used metal additive manufacturing processes. (a) Direct energy deposition (DED). (b) Laser powder bed fusion (L-PBF). (c) Electron powder bed fusion (E-PBF)[44].

### 3.1 Features

In laser powder bed fusion, metallic parts undergo complex thermal cycles [45, 46]. Initially, the powders interact with the laser beam and are melted at a fast heating rate. As the heat source moves away, the melt pool cools down rapidly and is solidified. The as-solidified parts then experience additional thermal cycles arising from the melting of subsequent tracks and neighboring layers.

#### (1) Rapid solidification

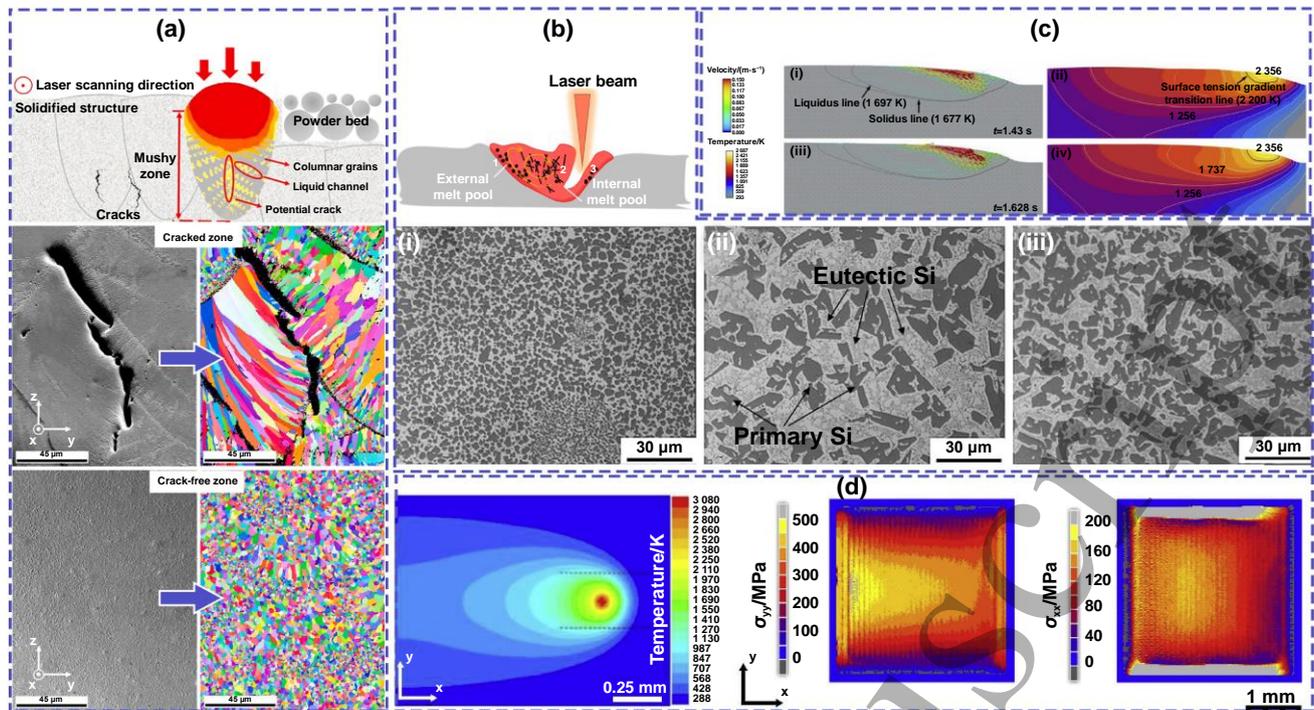
During L-PBF, the layer-by-layer scanning of high-power laser is essentially a non-equilibrium process [47]. A large amount of heat input will be applied to the metal powder, and the powder temperature can exceed  $2000^\circ\text{C}$  (Figure 2(c)) [48]. This results in a steep temperature gradient from the center of the melt pool to its boundary. Upon laser removal, the material rapidly solidifies at a cooling rate of roughly  $10^5 \text{ }^\circ\text{C}\cdot\text{s}^{-1}$ [49].

As a comparison, the cooling rate in conventional casting is approximately  $0.5 \text{ }^\circ\text{C}\cdot\text{s}^{-1}$  [43]. Consequently, the additively manufactured microstructure is one or two orders of magnitude smaller than the as-cast materials [50]. The repeated ultrafast heating and cooling cycles during L-PBF also gives rise to a higher dislocation density comparing to their cast counterparts [12]. Moreover, the additively manufactured microstructure exhibits supersaturated solid solution with micro-segregation along dendrite arms [51] as a result of the high cooling rate. Furthermore, the steep temperature gradient leads to an intense Marangoni flow in the melts [52], affecting the chemical uniformity within the melt pool [53].

## (2) Intrinsic heat treatment

In casting, materials typically undergo a single heating-cooling cycle. In contrast, in laser powder bed fusion, the powders are melted and cooled first, and the resulting solidified part can undergo multiple cycles of reheating or even remelting as the laser scans adjacent hatches or subsequent layers. The intricate physical process and thermal history result in high thermal residual stresses, heterogeneous metastable microstructures, and non-equilibrium compositions or phase distributions within the final part [20, 54, 55] (**Figure 2** (b) and (d)) [20, 48].

For instance, the intrinsic heat treatment during L-PBF enables the element partitioning at grain boundaries and potential precipitate formation. However, this unique feature of L-PBF can also bring manufacturing challenges for some specific alloys. The precipitates at grain boundaries, dendritic boundaries, and melt pool boundaries can cause boundary liquation or local stress/strain concentration due to the layer-by-layer scanning pattern during L-PBF. For instance, micro-segregation of Nb and Ti in L-PBF Inconel 718 alloy leads to the formation of brittle Laves and  $\delta$  phases, increasing the crack susceptibility. Homogenization heat treatment is thus generally employed to equalize element distribution (e.g., Ti and Nb in Inconel 718) and potentially dissolve harmful phases (e.g., Laves and  $\delta$  phases) present in the as-printed alloys [54]. Another typical characteristic of L-PBF is the high internal thermal stress, which is accumulated during thermal cycling. The thermal stress could cause cracks and geometric distortions of the components during L-PBF (**Figure 2**(a)) [56]. Furthermore, the L-PBF-processed alloys are inherently anisotropic [57] as a result of epitaxial grain growth, which is prevalent in laser powder bed fusion. Large columnar grain is known for the capillary effect and poor strain accommodation capability [48]. As a result of the microstructural non-uniformities [54], the mechanical response depends on the loading direction relative to the build direction, leading to anisotropic mechanical properties.



**Figure 2.** Effects of complex physical processes and thermal history of LPBF on microstructure, phase and stress distribution. (a) Schematic diagram of cracking in Al-Cu-Mg alloy and the microstructure of cracked and uncracked regions [56]; (b) L-PBF processed over-eutectic Al-50Si alloy. Complex thermal physical processes generate non-equilibrium phase distribution [20]; (c) longitudinal sections of mass flow velocity and temperature distribution of depositing single track at different times [48]; (d) anisotropic stress field in each layer under the influence of multiple anisotropic beam paths[55].

### 3.2 Defects

In L-PBF, the quality of printed parts is influenced by multiple factors such as metallic powders (flowability, particle size distribution, et al.), laser beam (beam diameter, beam intensity profile, et al.), and scanning strategies (hatch manner, inter-layer rotational angel, et al.). The defects occurred in additively manufactured alloys primarily include gas pores, keyhole, lack-of-fusion pores, and cracks, as shown in **Figure 3** [58].

The **gas pores** in the as-built microstructure typically originate from either the gas pores in the feedstock powders or the shielding gas entrapped during intensive melt pool flow in L-PBF. The gas pores primarily occur at occasions when there is insufficient time duration for the gas bubble to escape from the melt pool prior to solidification.

**Lack-of-fusion** (LoF) pores, as the name implies, are due to the insufficient melting of powders and are evident to occur at low energy input. The un-completely melted powders constantly leave a triangle space between two neighboring hatches. Due to the sharp appearance, the LoF is detrimental to the mechanical behavior.

In contrast to LoF, which occurs at insufficient energy input, the **keyhole** defect is observed at excessive

1 heat input. During L-PBF, the high-energy laser beam interacts with the powder beds. The powders at the  
2 surface boil and form a strong steam jet as a result of local heating. The steam jet then pushes the melt  
3 down due to the recoil pressure, forming a vapor depression (**Figure 3**), which is known as keyhole [59].  
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7 The keyhole, gas porosity, and LoF defects (**Figure 4** (a)-(c)) can be minimized or completely removed  
8 in the as-built parts by optimizing the scanning parameters or applying hot isostatic pressing (HIP)  
9 treatment. However, cracks, which are intolerant in most of the LPBF-processed parts, are still intractable  
10 for the majority of the existing metals and hinder the application of L-PBF in the industry.  
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15 Depending on their origins, cracks occurred during L-PBF are categorized as hot cracking and solid-  
16 state cracking [60]. Hot cracking includes solidification cracking and liquation cracking. Solid-state  
17 cracking involves strain-age cracking and ductility-dip cracking.  
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23 **Solidification cracking**, also termed as hot tearing during welding, is believed to occur at the last stage  
24 of solidification [61]. Due to the rapid cooling, solute atoms are enriched in-between adjacent dendrites,  
25 reducing the solidus of the residual liquid. The shrinkage strain caused by solidification and thermal  
26 contraction can pull apart the residual liquid. If the liquid is insufficient to backfill the inter-dendritic  
27 cavities, cracks form with clear dendritic characteristics [62] (**Figure 4** (d)-(f)). It is widely accepted that  
28 the semi-solid zone is prone to solidification cracking [63]. Therefore, narrowing the solidification  
29 temperature range results in a small slurry zone and a reduced hot cracking susceptibility.  
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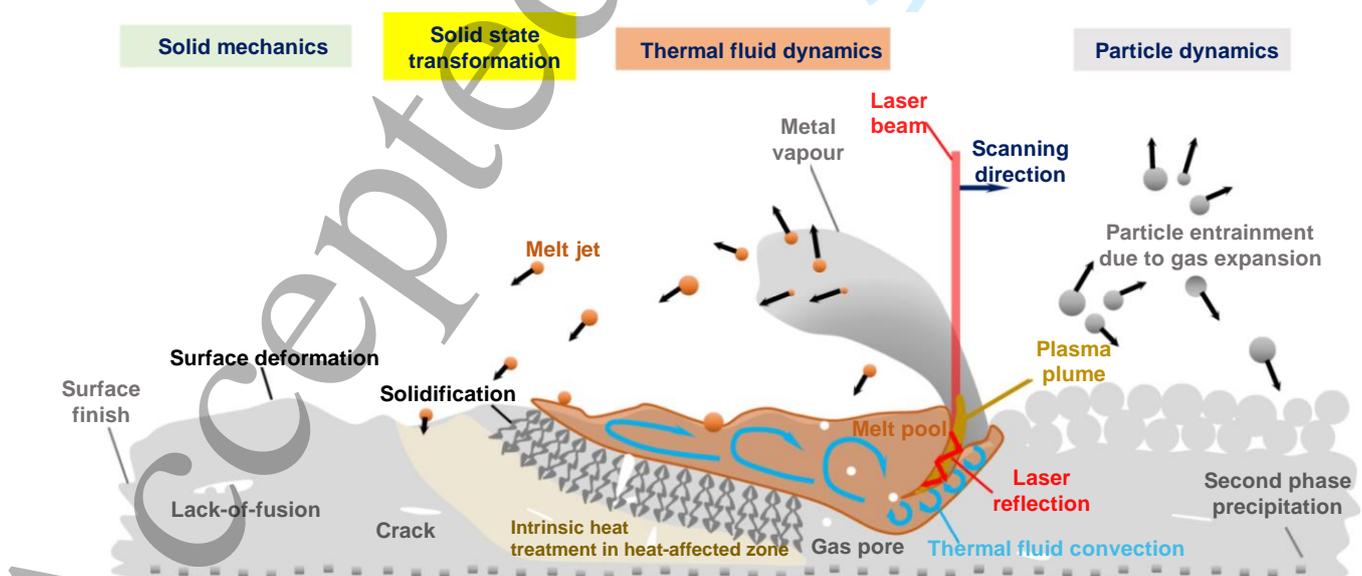
38 **Liquation cracking** occurs when the low melting point phase present in a solidified part is liquefied  
39 again due to the reheating from the adjacent regions (**Figure 4** (g)-(i)) [64]. In conventional welding, the  
40 liquation cracking is observed at the heat affected zone. During L-PBF, the liquation cracking generally  
41 occurs near the melt pool boundaries, which is attributed to the segregation induced liquation under the  
42 effect of cyclic heat treatment [65]. The enrichment of solute, e.g., boron [18] in superalloys, at inter-  
43 dendritic region significantly reduces the local solidus temperature, promoting the formation of liquation  
44 cracking.  
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53 **Strain-age cracking** (SAC) is mostly observed in  $\gamma'$  strengthened nickel-based alloys (**Figure 4** (j)-(l))  
54 [64]. The SAC cracking is attributed to the joint effect of  $\gamma'$  precipitation and the residual stresses during  
55 L-PBF. The  $\gamma'$  phase might precipitate during the layer-by-layer printing, leading to local stress  
56 concentration. The concentrated stress is superposed with the large thermal stress accumulated during L-  
57 PBF, leading to strain-age cracking.  
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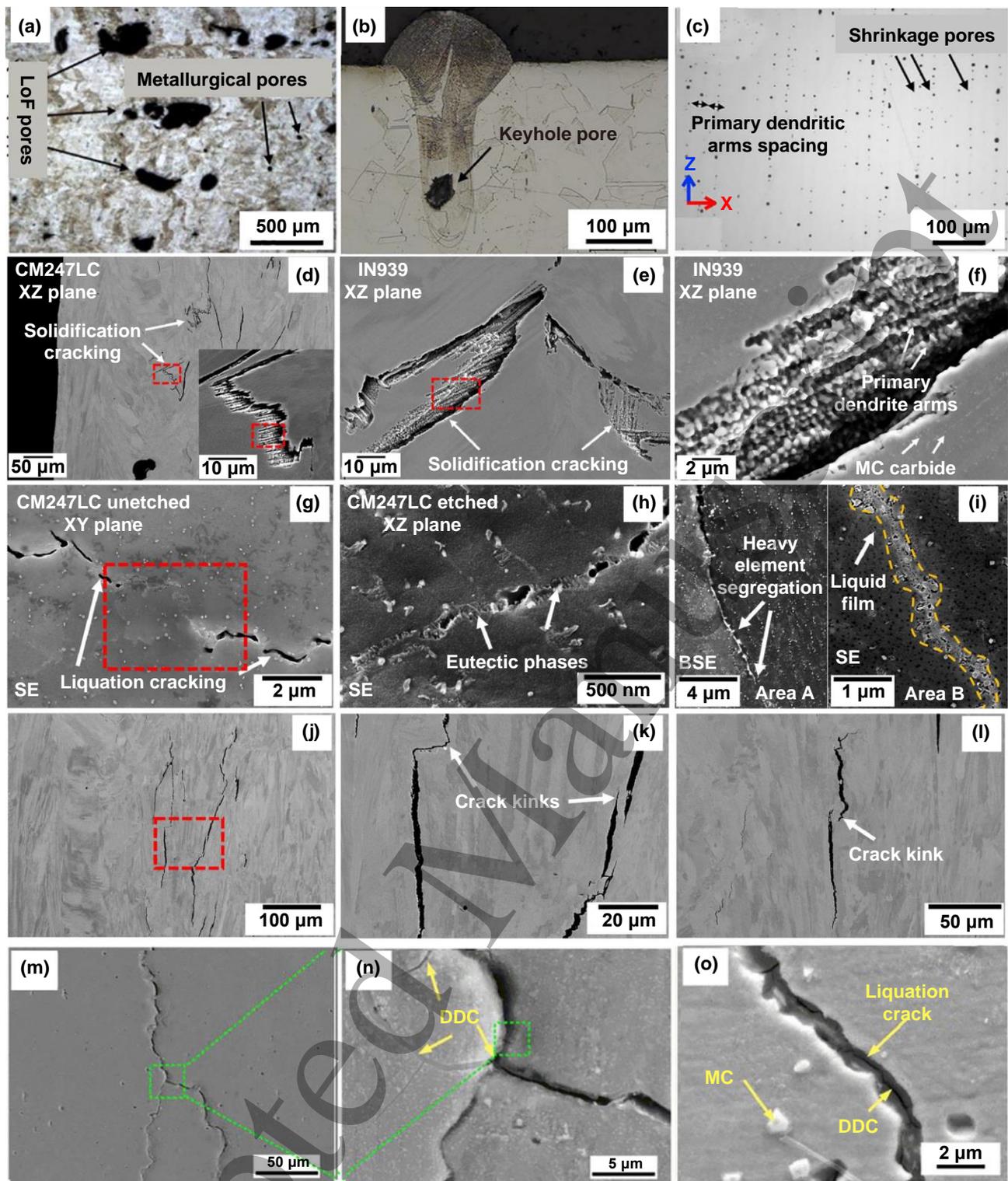
The mechanism of **ductility-dip cracking (DDC)** has not yet been completely understood. The grain boundary inoperability at the intermediate temperature range was found to be one of the factors causing DDC [66-68]. It is found that the ability of grain boundary sliding is significantly reduced at tortuous grain boundaries or at grain boundaries where precipitates like  $\gamma'$  and carbides are present. The suppressed grain boundary sliding is beneficial for enhancing creep resistance, yet, it causes strain concentration and generates voids. The connection of these voids significantly lowers the ductility and eventually leads to ductility dip cracking (**Figure 4 (m)-(o)**) [64].

Both processing parameters and alloy chemistry are found to have a significant impact on the cracking behavior. However, no consensus on the root cause of cracking has been reached yet. In general, the alloys with high solidification shrinkage (nickel-based superalloys) and thermal expansion coefficient (aluminum alloys) are susceptible to cracking during L-PBF [69]. Moreover, the refractory metals (e.g., tungsten, tantalum, and molybdenum) and metals with high laser reflectivity (e.g., pure copper) are less printable as they are difficult to melt [70]. The alloys with volatile elements (magnesium, zinc, etc.) are also challenging due to the server evaporation during L-PBF.

In the past, most of the work focused on the adaption of existing alloys to additive manufacturing process. However, the thermal circumstance, solidification process, the melt pool kinetics, etc. of L-PBF are distinct from those of the conventional processes. Only an integrated consideration of the alloy design and the unique features of L-PBF could lead to the synergistic advancement of new materials and processing technologies [71].



**Figure 3.** Complex metallurgical reactions during L-PBF [58].



**Figure 4.** Common defects and their macroscopic and microscopic morphology: (a)-(c) porosity [64], (d)-(f) solidification cracking [72], (g)-(i) liquation cracking [72], and (j)-(o) solid-state cracking [64].

#### 4. Printability of existing alloys

Defects, such as cracks in alloys, are intolerable for most of the applications. However, as mentioned in section 3.2, the existing alloys designed for conventional processes are not always suitable for the L-PBF process. As a result, defects such as pores and cracks can occur in alloys processed with L-PBF, and these defects are difficult to eliminate during the post-heat treatment. While hot isostatic pressing (HIP) can help

1 to close internal cracks and pores, it is not capable of closing surface cracks. Moreover, HIP is both costly  
2 and time-consuming, which is less favorable for alloy processing. Hence, it is crucial to design alloys with  
3 higher crack resistance to prevent cracks from occurring in the first place.  
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7 Similar to weldability, which is widely used in the welding industry to characterize whether an alloy is  
8 easier to weld than others, printability is employed in the additive manufacturing industry. It is the term  
9 rating the ability of an alloy to be printed without defects, particularly cracks[15]. The printability of an  
10 alloy is generally determined by trial-and-error tests. In the literature, there are several mathematical and  
11 empirical models assessing the cracking vulnerability of alloys during L-PBF. Most of the models or  
12 equations were modified based on the understanding of welding or casting processes. However, it is worth  
13 mentioning that additive manufacturing, particularly L-PBF, has much steeper temperature gradients and  
14 higher cooling rates than welding and casting, leading to large deviation in microstructure and cracking  
15 behavior. Therefore, it is essential to know whether the criterion for weldability is still valid for the L-PBF  
16 printability for designing easy-to-print alloys. In this section, the printability of the most commonly studied  
17 alloys, i.e., steels, nickel-based alloys, aluminum alloys, and titanium alloys, is evaluated with special  
18 attention on the comparison with weldability. The influence of each element on the crack formation is  
19 analyzed statistically, providing fundamental knowledge for alloy design.  
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#### 34 **4.1 Fe-based alloys**

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37 The Fe-based alloys, which have been extensively studied with L-PBF, include austenitic stainless steels  
38 (304L [73, 74], 316L [43, 75, 76], etc.), precipitation hardening steels (17-4PH [77, 78], 15-5PH [79, 80],  
39 etc.), carbon tool steels (H13 [81, 82], etc.), and maraging steels (300M [83, 84], etc.).  
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44 The research emphasis of AM'ed steel was mainly on property enhancement as fully dense steel parts  
45 seem to be easily manufactured [77, 78, 85]. The mechanical properties of these steels are comparable to  
46 their conventionally made counterparts. The printability of steels is fairly satisfactory. However, the fact is  
47 that only a handful of commercial steels can be printed without cracks, which is marginal compared with  
48 more than 3 500 commercial steel grades [86] in practical applications. Furthermore, behind the decent  
49 manufacturability, numerous parametric studies have been conducted to screen out the processing window  
50 in which successful deposition is obtained. It is of great value to examine the printability of the steel alloys,  
51 aiming at exploring the key elements leading to cracks during L-PBF.  
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In the welding industry, the term carbon equivalent (CE) was widely accepted as an indicator to quantify

1 the weldability of steels, as shown in **equation 1**.

$$CE = C + \frac{Mn}{6} + \frac{Cu+Ni}{6} + \frac{Cr+Mo+V}{5} \text{ (wt. \%)} \quad \text{Equation 1}$$

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7 According to the welding industry, steels with carbon equivalent above 0.35 wt.% have poor weldability  
8 [87]. Based on the steel composition, the carbon equivalent of each steel was calculated and displayed  
9 along with their printability in **Table 1**. Surprisingly, the high carbon equivalent value does not imply poor  
10 printability as expected. On the contrary, steels with unsatisfied printability, such as tool steels, have low  
11 carbon equivalent but high carbon contents. The low carbon equivalent in tool steels calculated with Eq.1  
12 is attributed to low Cr and Ni contents. This result demonstrates that the L-PBF printability is not merely  
13 linked to carbon equivalent, but also affected by the microstructure which is associated with all the elements  
14 in steels.  
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23 In addition to carbon equivalent, a Schaeffler diagram was widely used to estimate the phases of weld  
24 deposits based on the weldment composition, as shown in **Figure 5**. The Cr and Ni equivalents are  
25 introduced in the diagram, giving the proportions of austenite, martensite, and ferrite in the microstructure.  
26 As displayed in **Figure 5**, it is evident that the cracked steels during L-PBF mainly have martensitic  
27 microstructure, which is consistent with that observed in the weldment. Ferrite is found to resist hot  
28 cracking due to its high solubility of P and S [74]. Austenite has low solubility of P and S, thus facilitating  
29 the inter-dendritic segregation and leading to cracks [88]. Nevertheless, compared to ferrite, austenite has  
30 better ductility and a lower thermal expansion coefficient, which can reduce the risk of distortion and  
31 cracking during welding. Therefore, it is suggested to obtain ferrite-austenite bimodal phases at the end of  
32 solidification [74]. Compared with the planar grain boundaries in a single phase, the torturous inter-  
33 dendritic regions in ferrite-austenite solidification mode are more difficult for the liquid film to wet,  
34 suppressing crack propagation [74]. Compared with ferrite and austenite, martensite has limited weldability  
35 due to the high carbon content and the BCC crystal structure, which can lead to hard and brittle martensitic  
36 transformation during welding [89]. Preheating and post-weld heat treatment are generally applied to  
37 reduce the cooling rate and facilitate the formation of soft phases such as austenite, improving the ductility  
38 and toughness. As shown in **Figure 5**, the steels consisting of martensite are also more vulnerable to cracks  
39 compared to those with austenite and/or ferrite. This certifies that the effect of steel phases on weldability  
40 is consistent with that on the printability in L-PBF.  
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**Table 1.** The printability of Fe-based alloys during additive manufacturing.

Steel types	Steel name	Composition/(wt.%)									CE	Printability	Reference
		Cr	Ni	C	Si	Mn	Fe	Mo	Cu	Nb			
Austenitic stainless steel	304L	18.853	10.06	0.017	0.720	1.3	Bal	0.012	0	0	5.68	P	[74]
	316L	16.97	12.28	0.003	0.46	1.24	Bal	2.39	0.01	0	6.13	P	[43]
Martensite precipitation hardening steel	17-4PH	17.7	4.2	0.07	0.07	1	Bal	0	3.3	0.14	5.03	P	[77]
Martensitic stainless steel	15-5PH	14.65	5.01	0.043	0.61	0.578	Bal	0	3.83	0.368	4.54	P	[79]
	AISI431	16.34	1.85	0.15	0.53	0.48	Bal	0	0	0	3.81	NP	[90, 91]
	AISI420	13	<1	0.325	<1	<1	Bal	0	0	0	3.26	P	[85]
Maraging steel	14Ni-200	0.038	14.5	0.013	0.057	0.049	Bal	4.37	0.018	0	3.32	P	[92]
	L40	10.64	1.96	0.16	0.19	0	Bal	1.48	0.54	0	3.00	P	[81]
	Corrax	12	9.2	<0.05	0.2	0.2	Bal	1.4	0	0	4.30	P	[81]
Duplex stainless steel	18Ni-300	0.2	18.5	0.02	0.01	0.08	Bal	5.2	0	0	4.20	P	[83]
	AISI2205	22.88	5.45	0.028	0.333	1.818	Bal	3.105	0.224	0	6.47	-	[93]
	H13	5.125	<0.3	0.385	1	0.35	Bal	1.425	<0.25	0	1.85	NP	[81]
Tool steel	H11	5.125	0	0.4	1.025	0.4	Bal	1.35	0	0	1.76	NP	[81]
	M3:2	3.9	0	1.29	0	0	Bal	4.8	0	0	3.03	NP	[94]
	P20	1.95	0	0.4	0.45	0.83	Bal	0.33	0	0	0.99	-	[95]
	42CrMo4	1.05	0	0.415	<0.4	0.75	Bal	0.225	0	0	0.80	NP	[81]

To summarize, the printability of steels in L-PBF is influenced by key elements such as carbon and the phase constitutions. However, it should be noted that the carbon equivalent, which is commonly used to assess the weldability of steel alloys, is not valid in ranking the printability of the steels processed with L-PBF. The printability is also significantly affected by the steel microstructure. For instance, the presence of martensite in the as-printed microstructure can lead to higher susceptibility to cracking. This analysis suggests that conventional weldability may not always align with printability in L-PBF. The Schaeffler diagram is valuable in predicting steel printability with its chemistry. It is worth noticing that the Schaeffler diagram does not consider non-equilibrium solidification. During L-PBF, the elemental partitioning and the high thermal stress can induce unforeseen local phase transition, resulting in a more complex microstructure, deviating from that predicted in the Schaeffler diagram.

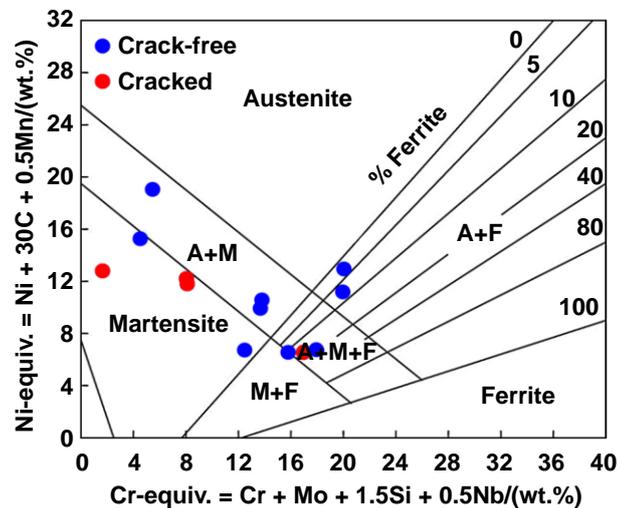


Figure 5. Schaeffler diagram with steel printability[96].

## 4.2 Ni-based alloys

Nickel-based superalloys generally have high strength, excellent creep resistance, and fatigue life at high temperatures [97]. They are widely used in severe high-temperature environments such as the marine and aerospace industries [98]. Conventional manufacturing of Nickel superalloys includes casting, directional casting, forging, and rolling. However, the more and more complex part structure challenges conventional manufacturing methods. A more geometry-adapted fabrication method is an urgent need. In recent years, L-PBF of nickel-based superalloy has been extensively studied and great progress has been achieved. For instance, more than 30 000 fuel nozzle tips have been successfully printed with Ni-based alloy by GE Aviation and used for LEAP engines[99]. However, the AM of nickel-based superalloys is still facing severe cracking problems. To date, only a limited amount of Nickel superalloys can be additively manufactured within a large crack-free window (the crack-free processing window is defined as the laser scanning parameter gap in which alloys can be printed without cracks and the scanning parameter can be input energy density), such as Inconel 718 and Inconel 625.

Ni-based superalloys can be strengthened by solid solution (Hastelloy X),  $\gamma'$  precipitates (CM247LC et al.), and  $\gamma''$  precipitates (Inconel 718 et al.). The alloys containing high fractions of  $\gamma'$  phase are known for their non-weldable behavior due to the high Al + Ti contents (>6.5 wt.%). Based on the welding industry, the nickel-based alloys containing Al + Ti > 4.5 wt.% are typically considered as non-weldable alloys [100], as shown in **Figure 6**. Haafkens *et al.* proposed a more generalized weldability assessment chart involving Cr and Co content, as shown in **Figure 7** [101]. The composition and printability of some nickel-based high-temperature alloys are shown in **Table 2**.

1 Considering that L-PBF is essentially similar to the welding process, it is interesting to know whether  
2 the weldability criterion is still valid for assessing printability. As shown in **Figure 6** and **Figure 7**, the  
3 cracked nickel alloys are scattered in both charts, particularly in **Figure 7** involving Cr and Co, manifesting  
4 that the printability of nickel-based superalloys is not always consistent with their weldability. For instance,  
5 the Hastelloy X (HX) and Haynes 230 were readily weldable, according to **Figure 7**. However, both alloys  
6 encounter severe cracking during L-PBF. Through efforts over the years, the crack-free Hastelloy X by L-  
7 PBF is eventually achievable (this is also the reason that HX is labeled as a printable alloy in **Figure 6** and  
8 **Figure 7**), but with strict control of minor elements. Tomus *et al.*[102] found that the HX powders with  
9 low Mn and Si contents could significantly reduce cracks during L-PBF. Later, they found that the Si and  
10 C contents played a major role in cracking mitigation as the two elements facilitated the formation of  $\sigma$   
11 phase and carbides (MC,  $M_6C$ ) due to the inter-dendritic segregation. Though the cracking mechanism of  
12 HX during L-PBF is still debatable, crack-free Hastelloy X powders are commercially available nowadays.  
13 Unlike HX, the cracking of Haynes 230 during L-PBF has not been completely solved without composition  
14 modification [103]. ABD900 and ABD850 are two newly developed nickel-based superalloys specifically  
15 for L-PBF [72]. Instead of having poor printability, as depicted in **Figure 7**, the two novel alloys are fairly  
16 processable during L-PBF without cracks. In the design of these two alloys, researchers considered the  
17 solidification temperature range, low melting point eutectic phases, and solid-state cracking.

18 The discrepancy between the printability and weldability of nickel-based superalloys is attributed to the  
19 unique processing characteristics associated with L-PBF, e.g., fast cooling rates, steep temperature  
20 gradients, cyclic heat treatment, and high thermal stresses. In conventional welding, higher Al + Ti content  
21 leads to an increased amount of  $\gamma'$  phase, while excessive Ti + Al can result in the precipitation of the  $\sigma$   
22 phase. The  $\sigma$  phase acts as nuclei for the formation of harmful topologically close-packed (TCP) phases,  
23 which can lead to welding cracks. However, in L-PBF, the rapid cooling rate significantly suppresses the  
24 precipitation, resulting in lower fractions of precipitates compared to what is expected based on the Ti + Al  
25 criterion. Therefore, the traditional weldability assessment criteria, i.e., Ti + Al content, may not directly  
26 apply to printability prediction of nickel-based superalloys in L-PBF.

27 In summary, the conventional metal manufacturing process considers the bulk composition and relatively  
28 low cooling rates, which enables substantial elemental distribution. However, during L-PBF, the extremely  
29 fast cooling rates combined with the intrinsic heat treatment lead to a complicated near-atomic-scale  
30 composition variation, deviating significantly from the bulk composition [18]. Therefore, the bulk criterion

for Ni-based alloy's weldability, e.g., (Ti + Al) %, is not always valid for the atomic-scale occasions during L-PBF. An in-depth understanding of the solute behavior at the atomic scale allows us to derive strategies to avoid cracking during laser powder bed fusion.

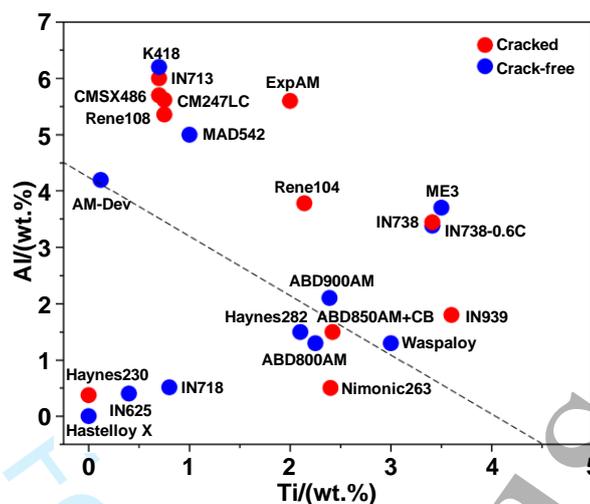


Figure 6. Correlation between printability and alloying elements in nickel-based superalloys.

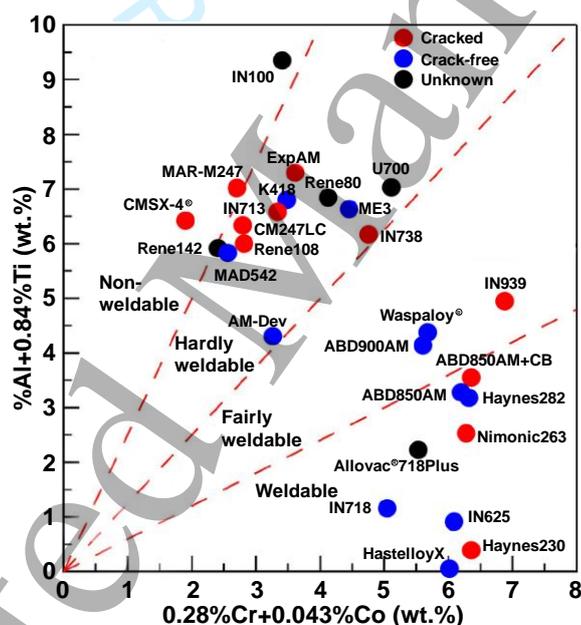


Figure 7. Effect of compositions on weldability of nickel-based superalloys. Modified weldability assessment chart after considering the effects of Cr and Co [104].

Table 2. The composition, printability and weldability of Ni-based superalloys.

Alloy	Composition/(w%)											Printability	Weldability	Reference
	Cr	Mo	Ti	Nb	Ta	W	Co	Al	C	Fe	Ni			
Inconel 718	18.5	2.91	0.8	5.11	0	0	0.01	0.51	0	17.76	54	P	W	[105]
Inconel 625	21.5	9.0	0.4	3.65	0.05	0	1	0.4	0.1	5	58	P	W	[106]
Inconel 738-0.6C	15.96	1.75	3.42	0.93	1.76	2.57	8.51	3.4	0.6	0	Bal.	P	W	[107]
Haynes 282	19.3	8.0	2.1	0	0	0	19.3	1.5	0.06	0	Bal.	P	W	[108]
Haynes 230	22.81	2.1	0	0	0	14.38	0	0.37	0.11	1.86	Bal.	NP	W	[109]
CM247LC	8.3	0.52	0.75	0	3.16	9.45	8.99	5.62	0.07	0	Bal.	NP	NW	[72]

Inconel 738	16.04	1.75	3.41	0.99	1.75	2.49	8.59	3.44	0.133	0.13	Bal	NP	NW	[110]
Inconel 939	22.3	0	3.6	1	1.4	2	18.8	1.8	0.15	0	Bal	NP	NW	[111]
Hastelloy X	21.2	8.8	0	0	0	0	2	0	0.06	17.6	50.04	P	W	[112]
Nimonic263	19.5	2.91	2.4	0	0	0	19.2	0.5	0	0.5	Bal	NP	W	[49]
Rene 108	8.64	0.53	0.75	0	3.02	10.03	10	5.36	0.01	0	Bal	NP	-	[113]
Rene 104	12.6	3.24	2.14	2.05	0.82	3.66	20	3.78	0.05	0	Bal	NP	-	[114]
CMSX486	5	0.7	0.7	0	4.5	8.6	9.3	5.7	0.07	0	Bal	NP	NW	[115]
ABD800AM	19.9	2.02	2.25	0.64	0.48	4.85	18	1.3	0.01	0	Bal	P	-	[72]
ABD900AM	17.1	2.09	2.39	1.85	1.21	3.06	20.3	2.1	0.047	0	Bal	P	-	[72]
ABD850AM+CB	19.7	2.03	2.42	0.37	0.57	5.05	18.6	1.5	0.133	0	Bal	NP	-	[116]
ExpAM	11.7	2	2	1.1	0	3.7	7.6	5.6	0.004	0	Bal	NP	-	[116]
Waspaloy	20	0.04	3	0	0	0	12	1.3	0.05	0.04	Bal	P	-	[116]
AM-Dev	8.7	1.2	0.12	3.6	5.5	7.1	19.1	4.2	0.03	0	Bal	P	-	[116]
Inconel 713	12	4.3	0.7	2.1	0	0	0	6	0.06	0.2	Bal	NP	NW	[116]
K418	12.5	4.3	0.7	2.1	0	0	0	6.2	0.12	1	Bal	P	W	[117]
MAD542	8	5	1	2	3	4	8	5	0.1	0	Bal	P	-	[63]
ME3	13.1	3.9	3.5	1.4	2.5	1.7	18.6	3.7	0.1	0	Bal	P	-	[63]

### 4.3 Al-based alloys

Aluminum alloys are known for their excellent strength-density compromise and are widely used in aerospace and automobile industries, in which both lightweight and mechanical properties are needed [15]. Laser powder bed fusion of aluminum alloys enables the design of geometrically complex parts with refined microstructure and enhanced mechanical performance [118]. Most Al-based alloys currently investigated in AM are directly adopted from those developed for the conventional cast and post thermo-mechanical processes but are not optimized for AM processing. Therefore, the intrinsic rapid solidification characteristics during beam-based AM processes raise significant challenges for the manufacturing of Al-based alloys, such as the high reflectivity, hot cracking, excessive loss of volatile alloying elements due to evaporation, porosity formation, and inadequate mechanical performance for structural applications that require high strength [5, 30, 34].

High reflectivity of Al and its alloys at a laser wavelength (around 1 064 nm) necessitates a high laser energy density during L-PBF and L-DED processing to achieve a dense part [119]. The high energy input during beam-based AM processing leads to the evaporation of versatile elements. Significant vaporization losses of alloying elements, such as Zn, Mg, Li, and Mn, have been widely reported in Al-Mn (3xxx), Al-Mg (5xxx), Al-Zn-Mg (7xxx), and Al-Li-based (8xxx) alloys during laser-based AM processing due to their low boiling points (907 °C for Zn, 1 090 °C for Mg, 1 342 °C for Li, and 2 061 °C for Mn) and high equilibrium vapor pressure (Zn > Mg > Li > Mn > Al > Fe > Ni > Ti) [120, 121]. In addition, Al-based alloys suffer from porosity issues during AM processing. The formation of excessive pores is mainly related to inappropriate process conditions (lack-of-fusion or keyhole pores), gas pores (vaporization of alloying

elements, entrapment of shielding gases or air), and hydrogen pores (moisture, grease, or hydrocarbon contaminants) due to the significant solubility difference of hydrogen in liquid and solid aluminum [122]. Hence, optimizing the process parameter and enhancing the protective environment help in minimizing the porosity defects.

Similar to that in welding, hot cracking is one of the most critical problems in AM processing of Al-based alloys. Majority of Al-based alloys required for structural applications in aerospace and automotive industries are prone to hot cracking during beam-based AM processing. The non-weldable 2xxx series Al-Cu-Mg-based and 7xxx series Al-Zn-Mg-based, particularly Al2024 and Al7075, are not printable during AM due to their high hot cracking susceptibility [120, 123]. During conventional welding, the 5xxx series Al-Mg-based and 6xxx series Al-Mg-Si-based alloys are susceptible to hot cracking or stress corrosion cracking depending on the Mg content [124]. Fortunately, the cracking can be readily prevented by using filler metals, such as Al-Mg-Mn (Cr, Ti) (Al5356), AlMg4.5Mn0.7 (Al5183), or Al4043 (AlSi5) [65]. However, the 5xxx and 6xxx series aluminum alloys suffer from hot cracking during AM processing where filler metals are not available [125-127]. It should be noted that manufacturing crack-free 2xxx series and 6xxx series aluminum alloys using L-PBF has proven to be possible by tuning the process conditions. For instance, hot cracking of Al-Cu-Mg-Mn can be avoided during L-PBF processing using a combination of low laser power (200 W) and low scan speed ( $83.3 \text{ mm} \cdot \text{s}^{-1}$ ) due to the enhanced liquid backfilling [128]. Crack-free AA6061 parts can be manufactured using high temperature baseplate preheating at  $500 \text{ }^\circ\text{C}$  during L-PBF due to significantly reduced thermal stresses [129, 130]. However, these strategies are not feasible solutions for widely industrial applications because the former sacrifices the production rate and the latter can cause substantial powder sintering and part oxidation. The current printable commercial Al-based alloys are limited to the 4xxx series, Al-Si-based alloys with a near eutectic composition, or 3xxx series Al-Mn-based alloys, for instance, AlSi7Mg, AlSi10Mg, and AlMn3 alloys [129, 130]. However, the low-to-medium mechanical properties of Al-Si-based and Al-Mn-based alloys could not meet the property requirements for structural applications in the aerospace and automotive industries. Therefore, there is a strong need to develop new high strength Al-based alloys tailored specifically to AM.

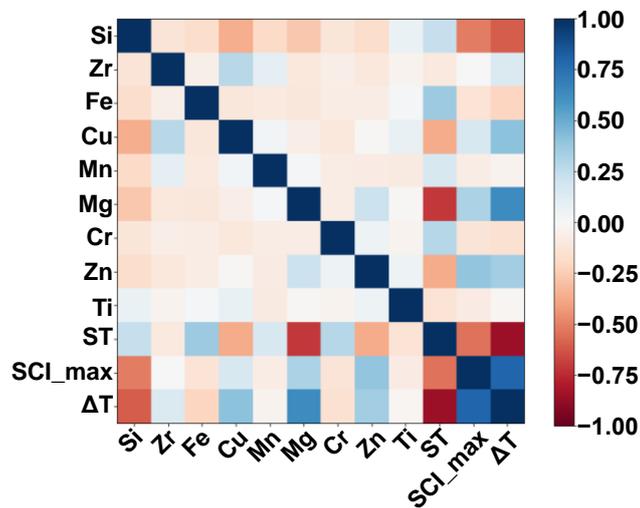
To design new Al-based alloys that can be processed crack-free using fusion-based AM technologies, it is crucial to identify the key influencing factors and their coupling effect on the cracking susceptibility. **Table 3** lists the chemical composition and printability during fusion-based AM processing of commonly used aluminum alloys. A Pearson correlation map was constructed to screen out the key factors responsible

1 for the cracking of aluminum alloys, as shown in **Figure 8**. The Pearson correlation coefficient is a  
2 statistical measure used to assess the strength and direction of the linear relationship between two variables.  
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4 It is calculated using the following formula:

$$\rho(x_1, x_2) = \frac{COV(x_1, x_2)}{\sigma_{x_1} \sigma_{x_2}} \quad \text{Equation 2}$$

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11 Here,  $\rho$  represents the Pearson coefficient,  $x_1$  and  $x_2$  are the two variables of interest, COV denotes the  
12 covariance between  $x_1$  and  $x_2$ , and  $\sigma(x_1)$  and  $\sigma(x_2)$  represent the standard deviations of  $x_1$  and  $x_2$ ,  
13 respectively. The Pearson coefficient ranges from -1 to 1, where 1 represents a perfect positive correlation,  
14 -1 indicates a perfect negative correlation, and 0 signifies no linear correlation. By calculating the Pearson  
15 correlation coefficient, one can gain insights into the interrelationship between variables and assess the  
16 strength and direction of their association.  
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24 In this study, the Pearson correlation diagram was generated using the SciPy module in Python,  
25 incorporating data from **Table 3**. **Table 3** presents the calculated values for three features: solidus  
26 temperature ( $ST$ ), brittleness temperature range ( $\Delta T$ ), and maximum solidification cracking index (SCI) at  
27 the brittleness temperature range. The SCI was calculated based on Kou's criterion [131], which was  
28 determined by the maximum slope of the  $T - (f_s)^{1/2}$  curve, where  $f_s$  is the solid fraction. Brittleness  
29 temperature range ( $\Delta T$ ) represents the temperature range when the solid fraction is between 0.8 and 0.99  
30 [132]. The three features were calculated using Scheil model in Thermo-Calc program with the TCAL6  
31 and MOBAL5 database. The default phase composition, a step size of 5 °C, an ambient temperature of  
32 25 °C, and a pressure of  $10^6$  Pa were used during the calculation. Once the Scheil solidification data was  
33 obtained, a Python program was written to process the exported data and calculate the three features. By  
34 examining the correlation between each element and these features, it is possible to identify the element's  
35 influence on the solidification process and its association with hot cracking. Note that this batch of alloys  
36 was not treated with the addition of secondary particles to eliminate hot cracking, as their crack elimination  
37 mechanism mainly involves grain refinement, which has little impact on the Scheil solidification curve.  
38 Therefore, they will be discussed separately in Section 5.2.3.  
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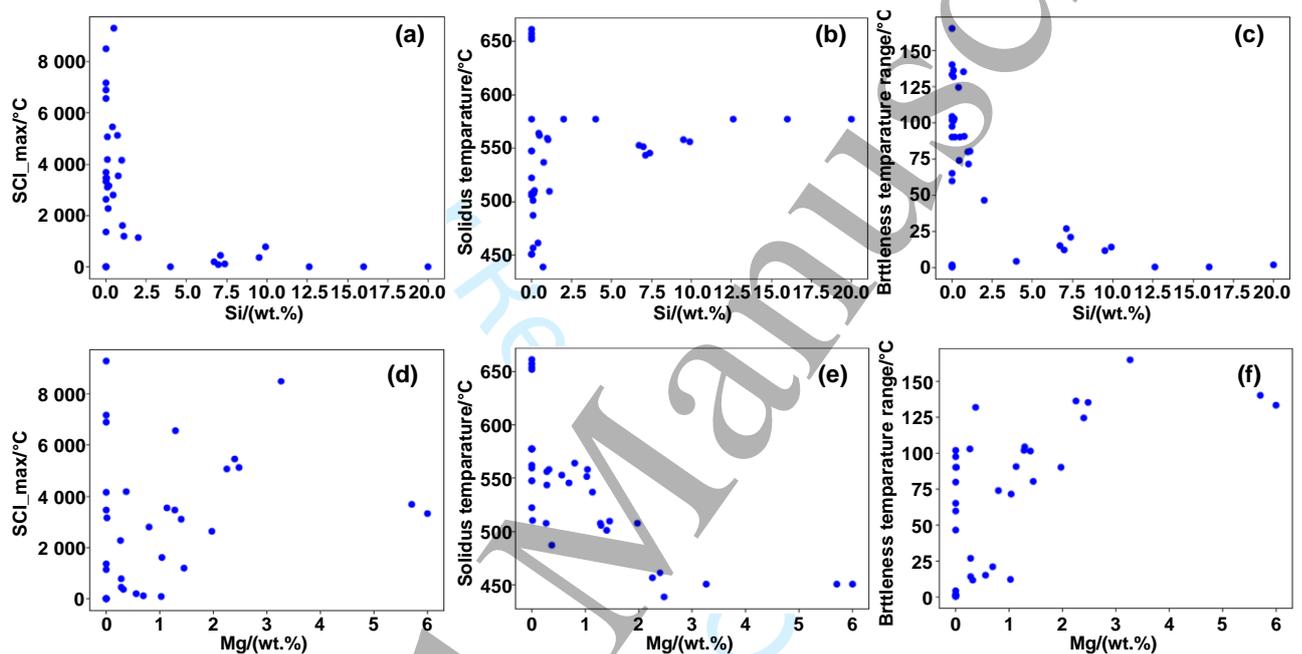
**Figure 8.** Pearson correlation between alloying elements and three features for aluminum alloys. *ST* represents solidus temperature,  $\Delta T$  denotes brittleness temperature range when the solid fraction is between 0.8 and 0.99, and SCI max indicates the maximum value of SCI in  $\Delta T$ .

As depicted in **Figure 8**, Si and Mg are the two elements that have strong correlations with the three features. Si positively correlates with solidification temperature and negatively correlates with the maximum SCI (SCI max) and brittleness temperature range ( $\Delta T$ ). It means that increasing Si content leads to a higher solidus and a lower SCI and  $\Delta T$ , thus reducing cracks. In contrast to Si, increasing Mg content results in a decreased solidus and increased SCI and  $\Delta T$ . This result implies that Mg should be lowered to reduce cracks during L-PBF. Furthermore, based on the Pearson correlation, Mg has the strongest linear correlation with ST, followed by Zn and Cu.

The specific correlations between key alloying elements on the three features were plotted in **Figure 9**. Increasing Si content raises the solidus temperature but reduces the SCI and  $\Delta T$ . Higher Mg content results in a lower solidus temperature, and higher  $\Delta T$  and SCI, increasing crack vulnerability, which is consistent with the Pearson correlation. Based on the statistical analysis, increasing Si and decreasing Mg are beneficial for crack mitigation for L-PBF-processed aluminum alloys. This is also in line with the experimental observations. Hyer *et al.* [133] investigated the effect of Si concentration on the cracking of Al-Si binary alloys processed via L-PBF. The results demonstrated that higher concentrations of Si in the binary Al-Si alloys exhibited lower cracking susceptibility. Moreover, Si is present as fine Si particles at low Si concentration and a web-like network of Al-Si eutectic at high Si concentration. An increase in Si concentration results in a eutectic network structure, which can act as a barrier to dislocation motion, enhancing the alloy strength. Li *et al.* [134] studied the Al-xMg-0.2Sc-0.1Zr ( $x = 1.5, 3.0, 6.0$  wt%) alloys and found that the crack density during L-PBF increased with the Mg content. However, when 1.3 wt. % Si was added, the hot cracking disappeared due to the significantly refined microstructure and the formation

of interdendritic Al-Mg<sub>2</sub>Si.

To summarize, the aluminum alloys typically exhibited poor printability and weldability. This is primarily attributed to the large solidification temperature range, which increases the hot crack sensitivity. Furthermore, the coefficient of linear expansion in aluminum is higher compared to the other three alloy systems [135], which could induce large thermal stress and increase crack vulnerability. Pearson correlation map helps screen out the decisive factors for cracking during L-PBF of aluminum alloy among a large number of elements. Big data and statistical analysis provide a potential methodology to discover the hidden correlation between alloy chemistry and cracking susceptibility.



**Figure 9.** Effect of (a)-(c) Si and (d)-(f) Mg contents on SCI (a, d), solidification temperature (b, e), and brittleness temperature range ( $\Delta T$ ) (c, f) of aluminum alloys.

**Table 3.** The solidus temperature, maximum SCI, brittleness temperature range, and printability of commonly seen aluminum alloys.

Alloy	Composition(wt.%)											Solidus temperature /°C	Maximum SCI/°C	Brittleness temperature range/°C	Printability	Reference	
	Al	Si	Zr	Fe	Cu	Mn	Mg	Cr	Zn	Ti	Ni						Li
AlCu2	Bal	0	0	0	2	0	0	0	0	0	0	0	547.5	6 895.8	97.6	NP	[136, 137]
AA2022	Bal	0.15	0	0.2	5	32	27	0.05	17	0.15	0	0	507.5	2 263.7	103.3	NP	[138]
AA2024	Bal	0.06	0	0.13	3	9	41	0	0	0	0	0	507.4	3 472.1	102.0	NP	[123]
AA2139	Bal	0.10	0	0.20	4	20	40	0	0	0	0	0	500.9	3 092.0	101.5	NP	[139]
AA2195	Bal	0	0	0	4	0	0	0	0	0	0	1	487.1	4 188.7	132.0	NP	[140]

1		al	12	13	00	38					0					
2											0					
3		B	0.	0.	6.	0.	0.	0.	0.	0.						
4	AA2219	al	2	17	3	3	02	0	1	05	0	0	510.3	3 169.0	90.2	P [141]
5				5						1						
6	AlCu6.8	B	0	0	0	6.										
7		al				8	0	0	0	0	0	0	547.6	1 368.2	59.6	P [136, 137]
8	AlCuMg	B	0	0	0	4.	0.	1.								
9	Mn	al				24	56	97	0	0	0	0	507.7	2 633.2	90.2	P [128]
10	AlCuMg	B	0	0	1.	2.	0	1.		0.	1.					
11	Ni	al			2	5	0	3	0	0	07	2	506.0	6 564.9	104.3	P [142]
12	Al-Cu-	B	1.	0	0	3.	0.	1.								
13	Mg-Si	al	12			56	62	45	0	0	0	0	509.7	1 206.5	80.5	P [143]
14																
15	Al-Cu-Li	B				4.										
16		al	0	0	0	0	0	0	0	0	0	2	522.1	3 467.8	102.2	NP [144]
17						0						5				
18	AlMn3	B	0	0	0	0	3	0	0	0	0	0	657.2	2.4	0.2	P [129]
19		al														
20	AlCr3	B	0	0	0	0	0	3	0	0	0	0	661.0	0.7	1.4	P [129]
21		al														
22	AlFe3	B	0	0	3	0	0	0	0	0	0	0	654.2	2.4	0.2	P [129]
23		al														
24	AlFe2.5	B	0.	2.												
25		al	03	0	55	0	0	0	0	0	0	0	652.0	0.5	2.1	P [145]
26	A356	B	7.	0.	0.	0.	0.	0.	0.	0.	0.	0.				
27	(AlSi7Mg	al	12	03	16	01	00	29	0	03	00	0	543.5	454.9	27.0	P [146]
28	0.3)					5				5						
29	A357	B	7.	0.	0.	00	0	0.		0.	0.					
30		al	37		15	5	7			5	5		545.5	111.5	21.2	P [147]
31																
32	F357	B	7	0	0.	0.	0.	1.		0.	0.					
33		al	7	0	15	05	03	02	0	05	2	0	551.4	101.5	12.3	P [148]
34								5								
35	AlSi7Mg	B	6.	0.	0.	0.	0.	0.	0.	0.	0.					
36	0.6	al	7	0	08	02	00	57	0	00	1	0	552.5	209.5	14.9	P [149]
37					5	5				5						
38	AlSi10M	B	9.	0	0.	0.	0.	0.		0.	0.					
39	g (CL31)	al	5	0	27	0	22	32	0	0	07	0	557.8	357.0	11.7	P [69]
40					5	5				5						
41	AlSi10M	B	9.	0	0.	0.	0.	0.		0.	0.					
42	g	al	92	0	13	0	00	29	0	01	00	0	556.1	787.4	14.3	P [150]
43					7	4	1			6	4					
44	AlSi0.5	B	0.	0	0	0	0	0	0	0	0	0	562.2	9 278.5	90.2	NP [133]
45		al	5													
46	AlSi1	B	1	0	0	0	0	0	0	0	0	0	559.2	4 155.8	80.2	NP [133]
47		al														
48	AlSi2	B	2	0	0	0	0	0	0	0	0	0	576.9	1 131.7	46.4	NP [133]
49		al														
50	AlSi4	B	4	0	0	0	0	0	0	0	0	0	576.9	1.1	4.1	P [133]
51		al														
52	AlSi12.6	B	12	0	0	0	0	0	0	0	0	0	576.9	0.2	0.2	P [133]
53		al	.6													
54	AlSi16	B	16	0	0	0	0	0	0	0	0	0	576.9	0.2	0.2	P [133]
55																

1		al																
2	AlSi20	B	20	0	0	0	0	0	0	0	0	0	0	576.9	2.5	1.8	P	[151]
3		al																
4	AA5083	B	0	0	0.	0.	3.							450.6	8 480.5	165.3	NP	[127]
5		al			41	88	27											
6	AlMg5.7	B	0	0	0	0	5.							450.6	3 691.7	140.2	P	[136,
7		al					7											152]
8	AlMg6	B	0	0	0	0	6							450.6	3 333.1	133.7	P	[136,
9		al																152]
10	AA6061	B	0.	0.	0.	0.	1.	0.	0.	0.	0.			536.9	3 534.3	90.8	NP	[153]
11		al	78	0	28	21	09	14	24	06	11	1	0					
12	AA6063	B	0.	0.	0	0	0.							563.9	2 802.4	73.9	NP	[154]
13		al	45	0	25		8											
14	AA6182	B	1.	0.	0.	0.	1.							557.7	1 616.3	71.5	NP	[155]
15		al	02	18	11	0	79	04										
16	AA7050	B	0.	0.	1.	0.	2.	0.	5.					461.3	5 446.8	124.6	NP	[120]
17		al	4	0	11	56	26	4	25	54								
18	AA7075	B	0.	0.	1.	0.	2.	0.	5.					456.7	5 060.8	136.3	NP	[69]
19		al	13	0	17	54	02	25	19	4								
20	AA7075	B	0.	0.	1.	0.	2.	0.	5.					438.6	5 129.8	135.4	NP	[156]
21		al	72	0	19	64	05	26	6.	09	0	0						
22			7			3	48	3	52	5								
23	AlZn5	B	0	0	0	0	0	0	5	0	0	0		577.0	7 150.7	65.0	P	[136,
24		al																137]

#### 4.4 Ti-based alloys

Titanium alloys, possessing desired strength, biocompatibility, and excellent corrosion resistance, have been widely used in biomedical, aerospace, and defense industries. Depending on alloying elements, titanium alloys can be classified as  $\alpha$ , near  $\alpha$ ,  $\alpha - \beta$ , near  $\beta$ ,  $\beta$ -titanium, and metastable  $\beta$  titanium alloys [71]. Conventional titanium producing processes include forging, casting, and powder metallurgy. Machining is generally needed to achieve the final geometry of the parts. However, titanium alloys are known as hard-to-machine materials due to their low Young's modulus and thermal conductivity, hindering their wide application in diverse industries [157]. As L-PBF is a near-net shaping technology, manufacturing hard-to-machine materials such as titanium alloys has drawn extensive attentions.

The easy-to-print titanium alloys include CP-Ti, which is pure titanium, and Ti6Al4V. The additive manufacturing of CP-Ti initially faces the impurity problem, particularly in controlling the oxygen level as titanium can form strong bonding with oxygen. Recent studies demonstrated that the oxygen content can be well-controlled, and the density of CP-Ti can achieve 99% [158]. Another titanium alloy that has been extensively studied in the AM community is Ti6Al4V, which is a typical  $\alpha$ - $\beta$  alloy. The as-fabricated

1 Ti6Al4V has predominantly refined  $\alpha'$  martensite through the  $\beta$  to  $\alpha'$  martensitic transition, leading to a  
2 higher yield strength ( $>100$  MPa), but a lower elongation to failure (5% ~ 10%), compared to the wrought  
3 alloy [71]. Lu *et al.* [159] found that the tensile strength is determined by the content of fine  $\alpha'$  martensite  
4 and lath thickness, while plasticity is influenced by prior  $\beta$  grain size and lath length. Increasing  $\alpha'$  fraction  
5 and refining prior  $\beta$  grains and  $\alpha$  laths simultaneously improves strength and plasticity. Wang *et al.* [160]  
6 improved the ductility and mechanical isotropy by reducing the Al content in Ti6Al4V from 6 wt.% to 4  
7 wt.%. This is due to the activation of multiple slip modes and twins during solidification and deformation.  
8 Chen *et al.* [161] utilized blended powders of Ti6Al4V and 3 wt.% Fe particles during L-PBF. The as-built  
9 microstructures transit from  $\alpha'$  dominated microstructure to a nearly complete  $\beta$ -dominated microstructure,  
10 exhibiting high strength and enhanced ductility without post-heat treatments. Seunghee A *et al.* [162]  
11 obtained martensitic  $\alpha'$  and  $\alpha + \alpha' + \beta$  microstructures by utilizing two different laser scanning speeds. The  
12 critical cooling rate of fully martensitic transformation is in the range between 2 900 and 6 500  $^{\circ}\text{C}\cdot\text{s}^{-1}$ .  
13 Although Ti6Al4V is widely used due to its inherent resistance to in-vivo corrosion, favorable  
14 osteointegration properties, and excellent strength-to-weight ratio, the release of toxic vanadium (V) ions  
15 in the human body poses health risks [163]. Issariyapat *et al.* [164] investigated the Ti-Zr alloy as a  
16 substitute for Ti6Al4V. They successfully prepared high-density Ti-Zr materials with homogeneous Zr  
17 distribution through laser powder bed fusion (L-PBF) using pre-mixed blends of Ti 0 ~ 10 wt% ZrH<sub>2</sub>  
18 powders. The presence of Zr promoted grain refinement, resulting in a fine acicular grain structure.  
19 Increasing Zr content significantly enhanced the tensile yield strength, while maintaining acceptable  
20 ductility. Ti-6Al-7Nb alloy, which replaces V with Nb in its composition, exhibits a martensitic  $\alpha'$  phase  
21 hardened by dispersed precipitates. Chlebus *et al.* [165] studied the effect of L-PBF manufacturing strategy  
22 on the mechanical properties and microstructure of Ti-6Al-7Nb alloy and found higher tensile and  
23 compressive strength but lower ductility compared to conventional methods. Ti-24Nb-4Zr-8Sn (Ti2448) is  
24 a near  $\beta$ -Ti alloy with a low Young's modulus (42 ~ 50 GPa) [166]. Introducing porosity further reduces  
25 the Young's modulus [167], making it suitable for porous titanium structures that match the human skeleton  
26 (4 ~ 30 GPa), which can be produced using L-PBF [168].

27 As discussed above, different from nickel-based and aluminum alloys, studies on the crack formation  
28 and mitigation of titanium alloys during L-PBF are rarely reported in literature, suggesting a good  
29 printability of titanium alloys. Compared to nickel-based superalloys, titanium alloys have lower Young's  
30 modulus [169]. The low Young's modulus enables titanium alloys to accommodate larger strains during L-

PBF. Such large deformation of titanium releases the high residual stress accumulated during intrinsic heat treatment of L-PBF process, alleviating the stress concentration and thus cracking behavior. Therefore, fully dense titanium alloys are achievable by optimizing scanning parameters, as shown in **Table 4**. The laser powder bed fusion of titanium alloys mainly focuses on suppressing epitaxial grain growth of long columnar grains, enhancing mechanical properties, and designing lattice structures for energy absorption and biomedical applications. A molybdenum equivalent MoE is also widely used to analyze the phase constitutions of titanium alloys, as shown in **equation 3**. If  $0 \leq \text{MoE} < 5$ , the titanium alloy is rich in  $\beta$  phase. If  $5 \leq \text{MoE} < 10$ , the alloy is near  $\beta$  phase. The alloy is metastable in the case of  $10 \leq \text{MoE} < 30$ , whereas for  $\text{MoE} > 30$ , alloys are considered stabilized.

$$\text{MoE} = 1.0(\text{wt. \% Mo}) + 0.67(\text{wt. \% V}) + 0.44(\text{wt. \% W}) + 0.28(\text{wt. \% Nb}) + 0.22(\text{wt. \% Ta}) + 2.9(\text{wt. \% Fe}) + 0.28(\text{wt. \% Nb}) + 0.22(\text{wt. \% Ta}) + 2.9(\text{wt. \% Fe}) + 0.28(\text{wt. \% Nb}) + 0.22(\text{wt. \% Ta}) + 2.9(\text{wt. \% Fe}) \quad \text{Equation 3}$$

**Table 4.** The composition, types, MOE and printability of Ti-based alloys.

Alloy	composition(w%)											Classification	Molybdenum equivalent MoE	Printability	Reference	
	Ti	Al	V	Nb	Mo	Cr	Fe	Zr	C	Si	Ta					Sn
Ti6Al4V	Bal	5.77	4.15	0.04	0.07	0	0.34	0	0	0.02	0	0.08	$\alpha+\beta$	9.62	P	[170]
Ti6Al7Nb	Bal	5.5	0	6.8	0	0	0.25	0	0	0	0	0	$\alpha+\beta$	8.13	P	[165]
Ti2448	Bal	0	0	24	0	0	0	4	0	0	0	8	Near $\beta$	6.72	P	[171]
Ti1023	Bal	3	10	0	0	0	2	0	0	0	0	0	Near $\beta$ / Metastable $\beta$	15.50	-	[171]
Ti5553	Bal	5	5	0	5	3	0	0	0	0	0	0	Near $\beta$	18.15	P	[171]
Beta C	Bal	3	8	0	4	6	0	4	0	0	0	0	Metastable $\beta$	21.96	-	[171]
Alloy C	Bal	0	35	0	0	1	0	0	0	0	0	0	Stable $\beta$	47.45	-	[171]
Ti-13Nb-13Zr	Bal	0	0	12.66	0	0	0	13.12	0	0	0	0	$\beta$ -rich	3.54	-	[172]
Ti-35Nb-5Ta-7Zr	Bal	0	0	34.5	0	0	0	6.5	0	0	4.5	0	$\beta$	10.65	-	[173, 174]
Ti-12Mo-6Zr-2Fe	Bal	0	0	0	12	0	2	6	0	0	0	0	$\beta$	17.80	P	[175]
Ti-5Al-5Mo-5V-1Cr-1Fe	Bal	5.2	5	0	4.93	1	0.96	0.01	0.01	0.02	0	0	Near $\beta$	18.02	P	[176]

1	Ti-6242	B	6	0	0	1.	0.0	3.9	0.0	0	0	1.	Near $\alpha$	8.07	P	[177]
2		al			92	0	5	3	1			89				
3	Ti-6246	B	6	0	0	6	<0.	4	<0.	0	0	2	Near $\beta$	12.00	-	[178]
4		al				0	15		04							

As discussed above, titanium alloys generally have fewer cracking problems during L-PBF comparing to Ni-based and Al-based alloys. This can be attributed to the smaller number of elements, which results in fewer low-melting-point precipitates at grain boundaries. Moreover, Ti-based alloys generally have a narrower solidification temperature range and a lower Young's modulus, which further reduces the hot crack sensitivity. Similarly, the weldability of titanium alloys is also quite satisfactory comparing to the other three alloy systems investigated in this work.

To summarize, while L-PBF offers numerous advantages, such as the ability to create complex geometries and produce functional metal parts, several specific issues can arise when using existing alloys in L-PBF.

(1) **Defects:** During the L-PBF process, rapid and localized heating and cooling cycles induce significant thermal stress in the printed part. This thermal stress can cause cracking and distortion, as existing alloys are not specifically designed to resist such stress during the forming process. Alloys with narrower solidification ranges tend to exhibit better printability, while those with large solidification ranges like CM247LC and AA7075 are more prone to cracking during L-PBF. Alloys with higher thermal conductivity can dissipate heat more efficiently, reducing the risk of thermal stress build-up and cracking. Mismatched thermal conductivities can lead to thermal gradients, warpage, and residual stresses, affecting part quality. Additionally, L-PBF requires high-quality powders as feedstock. Powder characteristics, such as normally distributed particle size, spherical morphology, and excellent flowability, are required to achieve desired print quality. However, not every conventional alloy can be processed into powders that meet these requirements. Powders with entrapped gas, undesirable morphology, or poor flowability can lead to defects such as porosity and lack-of-fusion in the printed part.

(2) **Compositions:** The alloy impurity in conventional processes can be reduced to extremely low levels using specific equipment like vacuum degassing furnace. However, in the powder-making process, the lack of specialized equipment or contamination from the refractory materials used during powder-making could increase the impurity levels in the powders. Additionally, the use of certain alloying elements with low melting point in conventional alloys, such as Mg and Al, can lead to issues like excessive powder spattering or vaporization during laser exposure. Furthermore, despite flushing the L-PBF equipment with inert gases

1 before printing, there might still be residual oxygen atoms present. This can result in oxidation of alloys  
2 that are prone to oxidation, which can degrade their mechanical properties. It is important to note that alloys  
3 highly reactive with oxygen, moisture, or other environmental gas, such as Mg, should be printed with  
4 specialized equipment or in controlled environments.  
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9 (3) **Microstructure:** Due to layer-wise build-up manner during L-PBF, the epitaxial grain growth is  
10 facilitated, leading to columnar grains along the building direction, which is undesirable for conventionally  
11 designed equiaxed grain structures. The rapid solidification during L-PBF can result in finer  
12 microstructures, but it can also lead to some metastable phases. The intrinsic heat treatment occurring  
13 during L-PBF can further contribute to the formation of undesirable phases, necessitating additional post-  
14 processing like heat treatment.  
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21 (4) **Post-Processing:** L-PBF parts often require post-processing steps such as heat treatment, hot isostatic  
22 pressing, or surface finishing to achieve the desired properties. However, these steps may differ from the  
23 conventional post processing optimized for casted or wrought alloys. Therefore, additional research and  
24 process optimization are required to develop post-processing techniques suitable for L-PBF fabricated parts  
25 using existing alloys.  
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31 (5) **Mechanical properties:** L-PBF fabricated parts exhibit anisotropic mechanical properties due to the  
32 layer-by-layer building process. The microstructure anisotropy might limit their performance in certain  
33 applications. Furthermore, the LPBF-processed traditional alloys might have reduced ductility and lower  
34 fatigue resistance due to the formation of fine-grained or non-equilibrium microstructures.  
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## 40 41 5. Alloy design strategies for L-PBF

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44 As discussed above, using existing alloys developed for conventional casting and forging processes in  
45 L-PBF can be challenging due to differences in solidification behavior, microstructure, properties, and post-  
46 processing requirements. To fully leverage the capabilities of L-PBF, it is crucial to develop alloys  
47 specifically designed for this technique.  
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53 Commercial alloys are designed for specific applications. The mechanical properties of commercial  
54 alloys have been certified and well-known. It is undesirable to modify the alloy composition as this would  
55 lead to deterioration of the mechanical performance. However, as discussed in section 4, the existing alloys  
56 are designed for traditional alloy-making processes without AM in mind. Consequently, server cracking is  
57 constantly observed in most of the commercial alloys processed with AM, particularly the L-PBF. Multiple  
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works attempt to modify the commercial alloy composition within the specification to improve printability. This is conservative as it is challenging to completely remove all the cracks in this way although it does not need to undergo an additional certification process. Some newly designed alloys achieve great success in crack mitigation, such as ABD900-AM [72], a novel nickel-based superalloy tailored for the L-PBF process. However, a knowledge-based alloy design guideline specifically for L-PBF is still absent. Three primary strategies are found in the literature. With mechanical properties in mind, one focuses on crack mitigation by studying the cracking mechanism, and the second tends to refine the grain structure. The third strategy employs the recently popular machine learning process without an in-depth understanding of the cracking or grain refining mechanism.

## 5.1 Crack mitigation-oriented strategies

Even in conventional casting, the mechanism of cracking is still under debate, although this topic has been studied extensively for years [179]. With the advance of AM, the attention on the cracking mechanism has been raised again. A lot of efforts have been spared on designing crack-free alloys for the L-PBF process. In this section, the crack criteria used during alloy design was presented at first. Subsequently, the grain and cell boundary manipulation work were reviewed as most of the crack-mitigation work focuses on the grain/cell boundary manipulations.

### 5.1.1 Cracking criteria

As discussed in section 3.2, cracks in LPBF-processed metals are classified as hot cracking, i.e., solidification cracking and liquation cracking, and solid-state cracking, which includes strain-age cracking (SAC) and ductility-dip cracking (DDC).

Various theories have been proposed to explain the formation of hot cracking during L-PBF, such as the RGD theory [180], Kou's criterion [131], solidification temperature range/freezing temperature range (temperature difference between liquidus and solidus), and brittleness temperature range (solid fraction between 0.8 and 0.99) [132].

Rappaz-Drezet-Gremaud (**RDG**) criterion for cracking is widely applied in L-PBF[180]. According to this criterion, hot tearing occurs if the pressure within the semi-solid mushy zone drops beyond a critical value. Besides, a viscous liquid is more difficult to flow and backfill the existing porosity or crack at the end of solidification. In the RDG model, the grain boundary can be categorized into repulsive (stabilization

of liquid films) and attractive boundaries (dendrite coalescence). The coalescence undercooling is found to be higher at high angle grain boundaries (HAGBs) than at low angle grain boundaries (LAGBs), indicating that the liquid film at HAGBs is stable at lower temperatures. This is consistent with the experimental observation that cracks are found to propagate along high angle grain boundaries, which is similar to welding [61].

**Clyne and Davis** [181] considered the time duration of the solidification intervals at high temperatures (between 10 ~ 60 vol.% liquid,  $t_r$ , indicating the time available for stress relaxation) and low temperatures (between 1 ~ 10 vol.% liquid,  $t_v$ , indicating the vulnerable time when cracks can propagate easily). The ratio of  $t_v/t_r$  is defined as the hot-cracking sensitivity (HCS). This model has succeeded in predicting the cracking susceptibility in aluminum alloys[180].

Xu *et al.*[63] proposed a heat resistance and deformation resistance (**HR-DR**) model for nickel-based superalloys processed via L-PBF. In this model, cracks are attributed to low heat resistance and deformation resistance at grain boundaries. Heat resistance (HR) is linked to the solidus temperature difference between the interdendritic (ID) and the dendritic core (DC) region. Deformation resistance (DR) is associated with the difference in the yield strength between DC and ID. According to the HR-DR model, alloying elements such as Zr, B, and Hf dramatically reduce the cracking resistance due to their strong segregation to interdendritic regions. Moreover, the  $\gamma'$  fraction has limited influence on the L-PBF as their precipitations are suppressed during the rapid cooling in L-PBF. The HR-DR model has been successfully validated using a novel nickel alloy MAD542 [182] and the commercial Rene 104 (also termed as ME3).

Another widely used cracking criterion was proposed by **Kou** [131], who considered the correlation amongst 1) separation of grains from each other due to solidification contraction, 2) the growth of two neighboring grains toward each other, and 3) the liquid backfilling between the intergranular channel. Based on these understandings, Kou derived an SCI to evaluate the hot cracking vulnerability of an alloy during solidification, as shown in **equation 4**.

$$SCI = \left| dT/d(f_s^{1/2}) \right| \quad \text{Equation 4}$$

where  $T$  is the temperature and  $f_s$  is the fraction of solid. The index was successfully verified with experimental data in casting and welding of Al alloys [131]. Additionally, Tang *et al.* [72] calculated the SCI of nickel-based superalloys, and found that the SCI of CM247LC, Inconel 939 and Inconel 738LC was

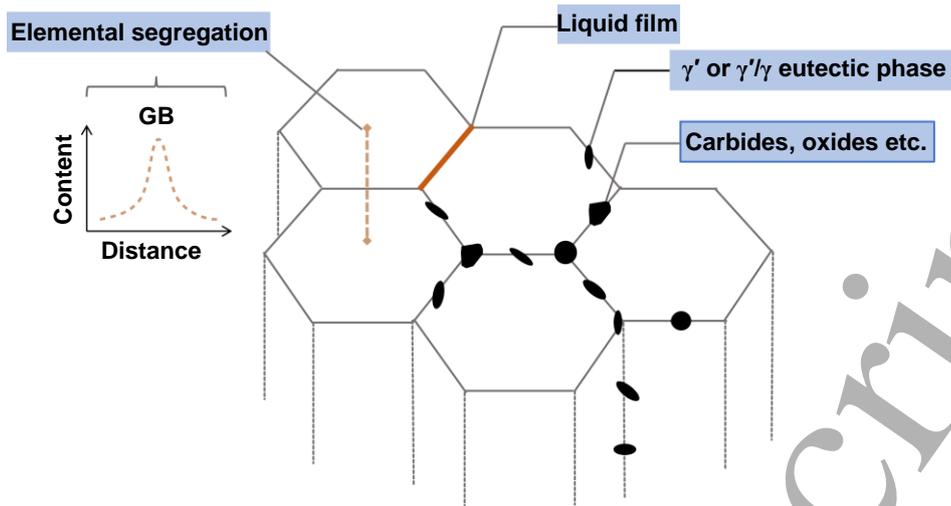
1 significantly higher than those easy-to-print alloys such as Inconel 718, which was consistent with the  
2 experimental observation.  
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5 **Solidification temperature range**/freezing temperature range has been widely employed to  
6 qualitatively assess the printability of alloys during L-PBF. Griffiths *et al.* [183] found that decreasing Hf  
7 significantly reduces the solidification temperature range of the nickel-based superalloy CM247LC. A Hf-  
8 free version of CM247LC was developed with crack density reduced by 36% ~ 75%. Tomus *et al.* [102]  
9 used two batches of pre-alloyed Hastelloy-X powders with different Si, Mn, and C contents for L-PBF. It  
10 was found that low Si and C contents reduced the solidification interval, helping to avoid crack formation.  
11 Therefore, by lowering Si and C content, crack-free Hastelloy X was fabricated via L-PBF. In terms of  
12 aluminum alloys, it is found that introducing silicon to aluminum alloys such as 2 024, 6 061, and 7 075  
13 can eliminate cracks due to the reduced solidification range, thermal expansion, and solidification shrinkage  
14 [184]. However, the solidification temperature range is not always valid in interpreting cracks. For instance,  
15 Zhao *et al.* [103] found that the addition of Zr (1 wt.%) in Haynes 320 resulted in a higher solidification  
16 temperature range. Instead of increasing hot crack sensitivity, the cracking was completely suppressed.  
17 Similarly, Sun *et al.* [185] introduced Al element into a high-entropy alloy CoCrFeNi, which was crack  
18 susceptible. Although the solidification temperature range was enlarged, adding 0.5wt.% Al significantly  
19 decreased the crack density. This was attributed to the grain boundary phase formation that lowered the  
20 tensile stress or even altered the stress state at the grain boundaries.  
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### 38 **5.1.2 Grain\cell boundary manipulation**

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41 The in-depth characterization of cracks demonstrates that grain boundaries and cell boundaries are the  
42 most vulnerable regions for cracking in L-PBF-processed microstructure. The rapid solidification during  
43 L-PBF leads to a supersaturated solid solution matrix. On top of that, the solutes that partition into the inter-  
44 dendritic liquid are insufficient to equilibrate via diffusion. This results in elemental enrichment at the  
45 solidification front based on the classical constitutional undercooling theory. The elemental enrichment  
46 assists the formation of intermetallics, carbides, or oxide inclusions, as shown in **Figure 10**. In casting, the  
47 intermetallic is beneficial for preventing hot cracking [186]. However, in the L-PBF process, the  
48 intermetallic at grain boundaries might be detrimental to the grain boundary integrities [187]. The  
49 occurrence of cracking during L-PBF is tightly associated with their boundary characteristics. Therefore,  
50 the concept of grain boundary segregation engineering was employed to manipulate the grain boundary  
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features to counteract the cracking problems in metal additive manufacturing [185].



**Figure 10.** Grain boundary features during laser powder bed fusion additive manufacturing.

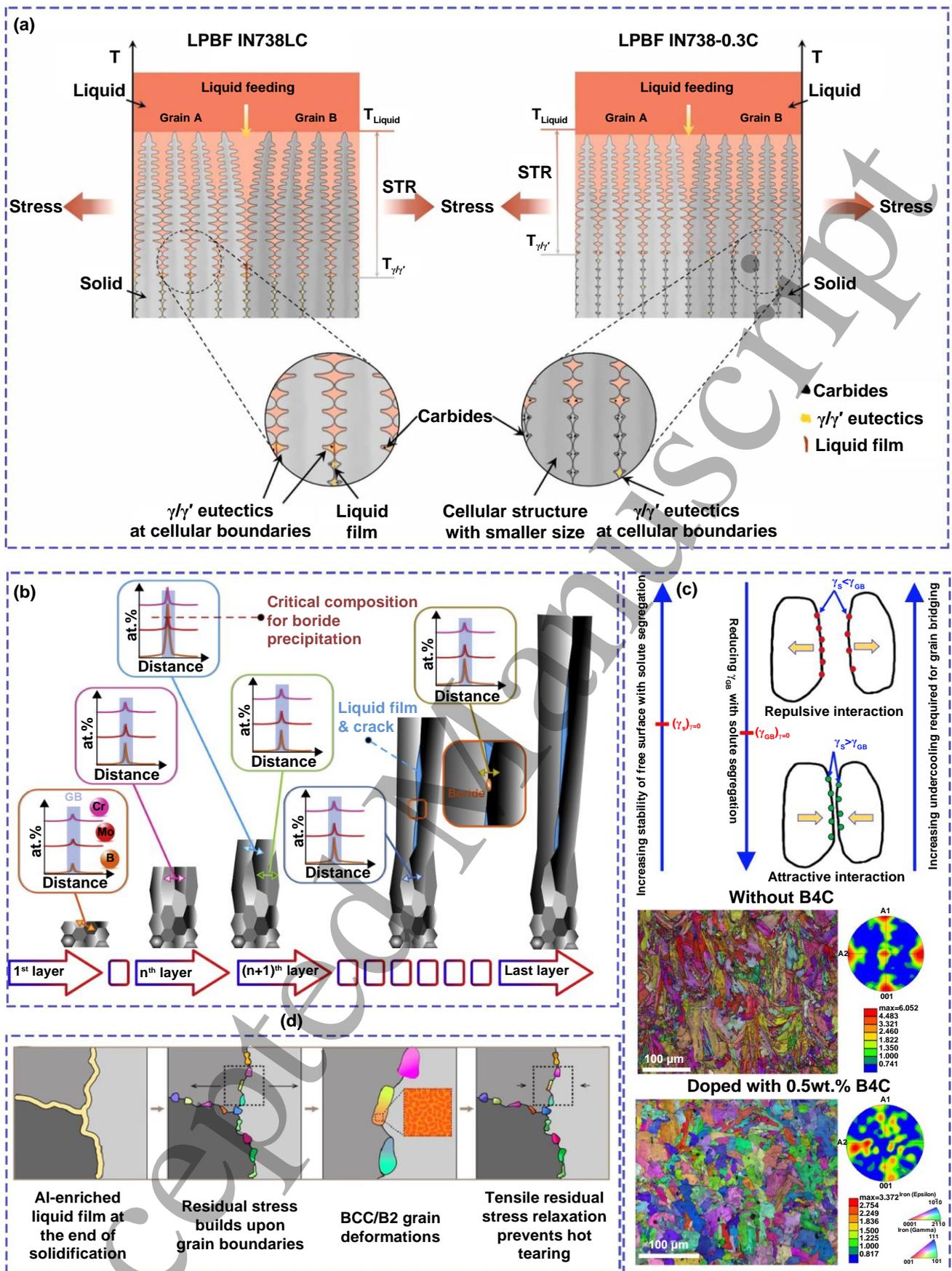
Sun *et al.*[185] found that the elemental segregation lead to a liquid film at the grain boundary at the end of solidification. Consequently, discontinuous precipitates formed along the grain boundary. The precipitates could accommodate the tensile residual stress through grain deformation and molar volume expansion. The tensile stress could thus switch into compressive stress, suppressing the formation of cracks, as displayed in **Figure 11(d)**. However, when the brittle precipitates form a continuous film, they promote crack initiation instead of acting as strain absorbers. Later, they developed a thermodynamics-guided approach for L-PBF alloy design using the concept of grain boundary segregation engineering [188]. The authors solved the hot cracking problem of Inconel 738LC alloy through lowering the Si content from 0.11 wt.% to 0.03 wt.%. They concluded that the occurrence of hot cracking during L-PBF required (a) pores to act as crack nucleation sources and (b) liquid films with a low solidus temperature to facilitate crack propagation, i.e., B in their work. Zhou *et al.*[107] found that with the carbon content increase from 0.11 wt.% (Inconel 738LC) to 0.3 wt.% and 0.6 wt.%, crack-free Inconel 738 was produced in a wide range of L-PBF parameters ( $76 \sim 126 \text{ J} \cdot \text{mm}^{-3}$ ). With the increase in carbon content, the  $\gamma/\gamma'$  eutectic decreased, and the cracking sensitivity decreased correspondingly (**Figure 11(a)**).

Kontis *et al.*[18] found that a segregated film at a HAGB can switch into a liquid film, as depicted in **Figure 11(b)**. The local thermal stresses and segregation-induced grain boundary liquation jointly cause hot cracking. Fewer grain boundaries lead to more concentrated and continuous liquid films extending over several millimeters, which is detrimental to mechanical performance. Therefore, they proposed to obtain an equiaxed or a columnar microstructure with a grain width smaller than  $100 \mu\text{m}$  to avoid cracking, despite

1 strong grain boundary segregation.  
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3 Thapliyal *et al.* [189] proposed a theoretical framework applying the concept of segregation engineering  
4 of grain boundaries during L-PBF. In order to use the proposed framework, the authors address the premises,  
5 which are: (a) certain elements must segregate at grain boundaries due to the low solid solubility, (b) solute  
6 segregation at the grain boundaries leads to reduced energy and mobility of the boundaries, and (c)  
7 segregation of the solute may lead to either an enhanced or deteriorated boundary cohesion depending on  
8 whether the solute is of strengthening or embrittling type (**Figure 11(c)**). They explored the concept to  
9 facilitate microstructural heterogeneity and hierarchy, suppressing cracking-susceptible columnar growth  
10 and leading to good printability at a wide range of process parameters.  
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20 Unlike most studies, in which researchers attempted to reduce the liquid film at grain boundaries, Zhao  
21 *et al.* [103] introduced liquid backfilling and a network of segregation phases at grain boundaries to  
22 alleviate thermal stress. Following this strategy, they added Zr in Haynes 230 and obtained crack-free  
23 specimens with an extraordinary combination of strength and plasticity.  
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**Figure 11.** Suppressing cracks during L-PBF using the concept of grain boundary segregation engineering. (a) Schematic of the inhibition effect of carbon on the cracking of L-PBF Inconel 738. The addition of carbon reduces the enrichment of element B at grain boundaries and decreases the formation of low-melting-point phases[107]. (b) Schematic of the microstructural and compositional evolution during AM of the superalloys with hot cracks. The progressive enrichment in Cr, Mo and B at grain boundaries over the building process causes grain

1 boundary segregation induced liquation [18]. (c) Schematic of solute segregation effect on interfacial energy and solidification behavior of  
2 metastable Fe-Mn-Co-Cr-Si high entropy alloy processed via L-PBF [189]. (d) Schematics of residual stress minimization for Al<sub>0.5</sub>CoCrFeNi  
3 alloy during L-PBF. The image indicates the Al segregation towards the interdendritic/grain boundaries, discontinuous BCC/B2 grains formed  
4 along the grain boundaries, the tensile residual stress accommodated by the BCC/B2 grains, and the switch from a large tensile residual strain  
5 a small compressive residual strain[185].  
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8 To summarize, the rejection of solutes during rapid solidification in L-PBF leads to cell/dendrite  
9 segregation. The enrichment of solutes in the residual liquid reduces the grain boundary energy and  
10 mobility and affects the grain boundary cohesiveness. Depending on whether the solute is of strengthening  
11 type or embrittling type, the grain boundary is then strengthened or easily cracked. Moreover, the  
12 segregation-induced grain boundary characterization can lead to the low-melting-point phase formation,  
13 which is not desired considering the hot cracking. Nevertheless, the precipitates can also reduce the  
14 detrimental effect of residual thermal strain and facilitate liquid backfilling, which are beneficial for  
15 cracking mitigation. Therefore, the grain boundary segregation should be carefully manipulated in order to  
16 design novel crack-free alloys for L-PBF.  
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## 26 **5.2 Microstructure manipulation-oriented strategies**

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28 In section 5.1, we summarized the work on designing alloys for L-PBF with the intention of removing  
29 cracks. However, crack mitigation is not the sole purpose of alloy design for L-PBF. In addition to cracks,  
30 the microstructure of LPBF-processed alloys also significantly impacts the mechanical properties,  
31 corrosion resistance and other functional properties, thus affecting the alloy's application. Generally, the  
32 layer-by-layer printing nature of the L-PBF process is favorable for the columnar grain growth. Columnar  
33 microstructures, ideally single crystalline, are desirable for applications such as gas turbine blades to  
34 enhance creep and fatigue properties. However, in most cases, microstructure anisotropy is undesirable,  
35 and a refined and equiaxed microstructure is preferred. Furthermore, due to the fast cooling rate, the as-  
36 built alloys are constantly a supersaturated solid solution. While this may be favorable for solid solution  
37 alloys, precipitation hardening alloys typically require post-heat treatment to achieve optimal properties.  
38 Therefore, when designing alloys for the L-PBF process, it is crucial to consider the microstructure  
39 preferences of the specific alloy. By accounting for the desired microstructural characteristics during alloy  
40 design, extensive parameter optimization work and post-heat treatment can potentially be avoided,  
41 simplifying the overall manufacturing process.  
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58 The high cooling rate and high thermal gradient of laser powder bed fusion almost exclusively lead to  
59 columnar grains, which are undesirable considering the anisotropic mechanical properties in most of the  
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1 applications [190]. According to the well-established grain refinement theory in casting, nuclei provide a  
2 site for growth, and solute leads to constitutional supercooling, which activates adjacent nuclei [191]. A  
3 growth restriction factor,  $Q$ , defined as the development rate of a constitutionally undercooled zone, is  
4 employed to assess the effect of an element on grain refinement [192]. Tan *et al.* [193] reconsidered the  
5 grain refinement theory and found that the classic theory (i.e.,  $Q$  value) could not be directly applied to  
6 metal L-PBF. The large thermal gradient during L-PBF could significantly inhibit the formation of a  
7 compositional supercooling zone ahead of the S/L (solid/liquid) interface. In addition, the solute rejection  
8 resulted in a lag in the actual growth of the S/L interface compared to the theoretical solidification rate. The  
9 growth restriction factor ( $Q$  value) cannot characterize the role of solutes in L-PBF. Despite the lack of  
10 exact grain refinement theories, a considerable amount of work has been reported in the literature, achieving  
11 refined microstructure in newly designed or tailored commercial alloys for L-PBF. Three different  
12 strategies were found in the literature to promote the columnar-to-equiaxed (CTE) transition during L-PBF,  
13 i.e., optimizing process parameters such as in steels [194, 195] and nickel-based alloys [196], the addition  
14 of solute elements with a high supercooling capacity like in titanium alloys [190, 197-200], and exploring  
15 grain refiners for aluminum alloys [201-206].

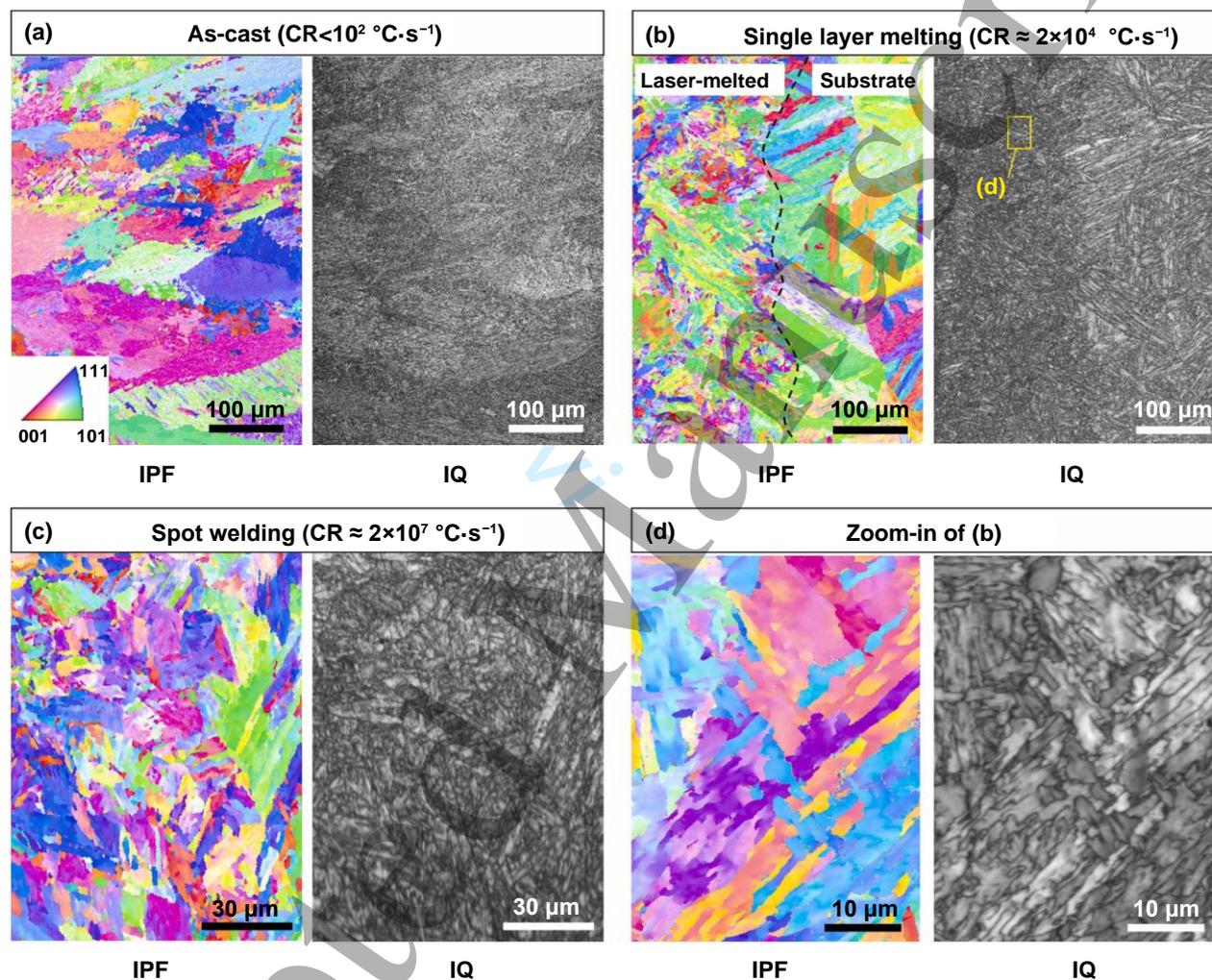
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32 In this section, the microstructure manipulation-oriented strategies were discussed, including phase  
33 transition, columnar-to-equiaxed transition, and grain refinement. Phase transition involves making use of  
34 the rapid cooling and intrinsic heat treatment in L-PBF to induce specific phase transitions, such as the  
35 precipitation of secondary phases. CET is employed to promote the formation of equiaxed grains through  
36 increasing the supercooling capacity. Grain refinement is realized by adding nucleating agents to increase  
37 the nucleation sites and promote the equiaxed grain formation.

### 38 39 40 41 42 43 44 45 **5.2.1 Phase transition**

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Due to the layer-by-layer printing nature and the complex thermal cycles during L-PBF, it is feasible to  
control the phase transition guided by thermodynamic calculation beforehand. Therefore, during alloy  
design for L-PBF, it is essential to take into account the potential phase transition that may occur during  
laser processing.

Guo *et al.* [194] proposed a phase-transformation-guided approach for alloy development in the L-PBF  
of PH17-4 steel. Through controlling scanning parameters, they minimized the formation of  $\delta$ -ferrite but  
facilitated the austenitization. Next, they removed the minor alloying elements C, Mn, and Si, which

reduced the  $M_s$  temperature, to promote the austenite-to-martensite transformation. In the end, UW\_17-4 steel was developed with a fully martensitic microstructure at various cooling rates (Figure 12). The authors address the importance of understanding the phase transformation dynamics prior to alloy design for AM. Wang *et al.* [195] fabricated 316L with L-PBF and obtained a super combination of yield strength and ductility, breaking the strength-ductility dilemma. It was interpreted that the unique cellular structures, low-angle grain boundaries, and dislocations formed during L-PBF resulted in the high strength, and the work-hardening mechanism contributed to the high elongation.



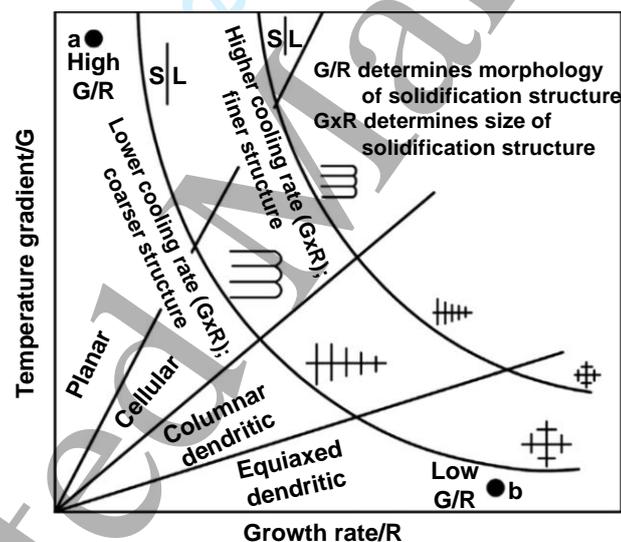
**Figure 12.** Microstructure of UW\_17-4 under different cooling rates. (a) EBSD of as-cast UW\_17-4 fabricated by arc-melting. The left panel is an inverted pole figure (IPF). The right panel is the corresponding image quality (IQ) map. (b) EBSD of UW\_17-4 after a single-layer laser melting (transverse cross-section). The substrate is a cast, fully martensitic UW\_17-4 after a solution heat treatment. (c) EBSD of UW\_17-4 after laser spot welding under 156 W laser power with 1 ms laser duration (transverse cross-section). (d) EBSD of a zoom-in area from (b). The microstructures for all conditions are fully martensitic. All IPFs share the same color code, which is shown in the inset of (a) [194].

The LPBF-processed alloy is constantly a supersaturated solid solution with a low yield strength and thus requires a post heat treatment. In order to avoid the post processing, in-situ precipitation by optimizing scanning parameters is proposed to induce nano precipitations under the intrinsic heat treatment during AM.

For example, Fe19Ni5Ti steel was tailored for laser additive manufacturing by local control of both nanoprecipitation and martensitic transformation [196]. The as-built steel was hardened in-situ by NiTi nanoprecipitation and it consists of alternating soft (without precipitates) and hard layers (with a high volume fraction of nanoscale precipitates) with a tensile strength of 1.3GPa and 10% elongation.

### 5.2.2 Columnar-to-equiaxed transition

During L-PBF, if the easy growth direction of crystals is parallel with the thermal gradient, epitaxial growth is facilitated. As a result, columnar grains spanning multiple melt pools are constantly present in the as-built microstructure. Variation in processing parameters during L-PBF determines the temperature gradient and cooling rates, thus affecting the crystal morphology and phase transformation. Although a lot of attempts have been made to promote the columnar-to-equiaxed transition (CET) during L-PBF by optimizing the scanning parameters as shown in **Figure 13** [65], it is still challenging to obtain fully equiaxed grains, particularly for titanium alloys [207]. Therefore, it is significant to design alloys with high CET tendency in order to achieve fully equiaxed grain structures.



**Figure 13.** The effect of temperature gradient  $G$  and growth rate  $R$  on grain size and morphology.  $S$  and  $L$  stand for solid and liquid, respectively[65].

Mereddy *et al.* [197] added different amounts of carbon to Ti6Al4V during additive manufacturing, and they found that the addition of 0.03 wt.% ~ 0.41 wt.% carbon reduced the prior  $\beta$  grain size and  $\alpha$  lath by factors of 5 ~ 6. The carbon segregated at grain boundaries during rapid cooling decreased the solidification temperature, leading to constitutional supercooling. Higher constitutional supercooling capacity thus suppresses the epitaxial growth, leading to a refined microstructure [192]. Gou *et al.* [198] proved that the addition of Nb in Ti6Al4V increased the supercooling capacity, and the planar grain growth mode switched

1 to the cellular grain growth, leading to a refined microstructure. Zhang *et al.* [190] developed Ti-Cu alloys  
2 that have a high constitutional supercooling capacity. The as-printed Ti-Cu alloys have a fully equiaxed  
3 microstructure without any complex processing parameter control. They demonstrated a pathway to  
4 additively manufacture Ti-Cu alloys with fine equiaxed prior- $\beta$  grains and an ultrafine eutectoid lamellar  
5 structure. Similarly, Narayana *et al.* [199] and Simonelli *et al.* [200] investigated the addition of Fe in  
6 Ti6Al4V alloy during direct energy deposition and laser powder bed fusion, respectively. They found that  
7 the addition of 3% ~ 4% Fe resulted in a significant reduction of prior  $\beta$  grains from ~190  $\mu\text{m}$  to ~70  $\mu\text{m}$   
8 [199] and the further increase of Fe (>4%) content showed only a limited further benefit [200]. Zhang *et*  
9 *al.* [208] found that adding boron to Ti6Al4V resulted in a reduced grain size and refinement of  $\alpha$  lath  
10 during additive manufacturing. Moreover, grain refinement weakens the texture, reducing the tensile  
11 property anisotropy.

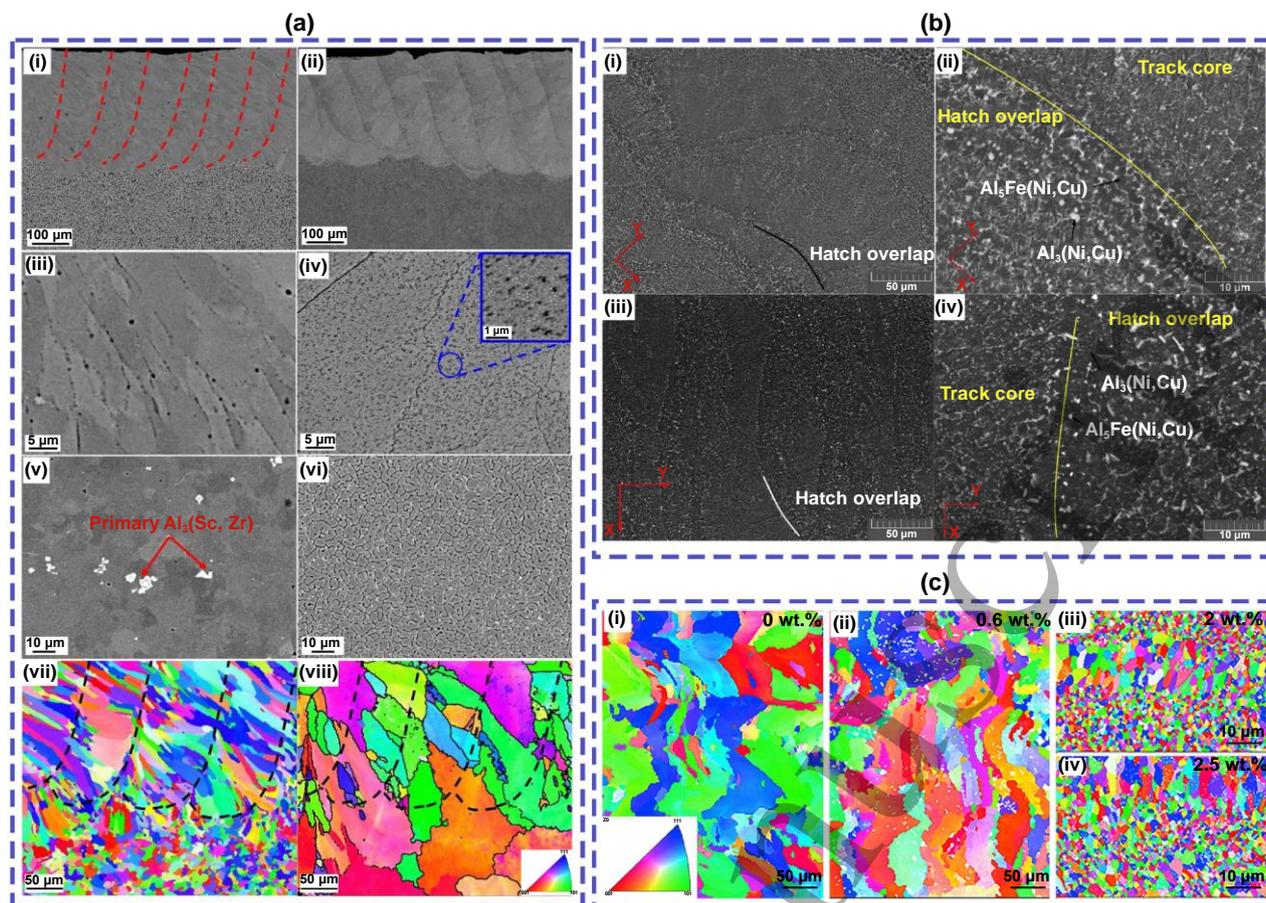
### 5.2.3 Grain refinement

12 Epitaxial grain growth, prominently prevalent in laser powder bed fusion, is known for the capillary  
13 effect and poor strain accommodation capability. Therefore, a lot of efforts have been made to refine grains  
14 to prevent hot cracking and enhance the mechanical property isotropy. Adding solute atoms as nuclei to  
15 promote equiaxed grain formation has been widely explored in aluminum alloys and titanium alloys [14,  
16 209-211]. However, in nickel-based alloys and steels, the inoculation treatment is not a common practice  
17 due to the lack of knowledge of the inoculants [103].

18 Researchers have attempted to add elements to the standard alloy as inoculants for grain refinement. A  
19 successful example is the Scalmetalloy, which is developed and patented by the Airbus Group [201]. The  
20 introduction of Sc or Zr element to aluminum alloys promotes the formation of a supersaturated solid  
21 solution under rapid cooling during L-PBF. The alloy was then strengthened by direct aging to form  $\text{Al}_3\text{Sc}$   
22 or  $\text{Al}_3\text{Zr}$ , which is an  $\text{L}_{12}$  structure and is strongly coherent with the Al matrix [202]. The  $\text{Al}_3\text{X}$  precipitates  
23 act as nucleating agents, leading to significant microstructure refinement for the aluminum alloys. The  
24 Scalmetalloy has both good printability and excellent mechanical properties, i.e., yield strength  $\text{YS} = 520$   
25 MPa, ultimate tensile strength  $\text{UTS} = 530$  MPa, and elongation = 14% [201]. Due to the success of  
26 Scalmetalloy with regard to both printability and mechanical properties, similar strategies were explored by  
27 other researchers.

28 Jia *et al.* [203] explored the potential of replacing Sc with Er in the Al-Sc-Zr alloy. They found that the

1 microstructure of the laser remelted Al-Sc-Zr region was mainly composed of fine columnar grains, while  
2 the remelted region of Al-Er-Zr alloy consisted of a large-grained structure (**Figure 14(a)**). The Al-Sc-Zr  
3 alloy has higher hardness and thermal ability compared with the Al-Er-Zr alloy. Manca *et al.* [204] printed  
4 the novel Al-3Ce-7Cu alloy via L-PBF with a low porosity of 0.03%. The mechanical performance is  
5 satisfactory with YS = 274 MPa and UTS = 459 MPa at room temperature. The high strength was attributed  
6 to the fine Al<sub>6.5</sub>CeCu<sub>6.5</sub> and Al<sub>11</sub>Ce<sub>3</sub> eutectic phase precipitated in the as-built Al-3Ce-7Cu alloy,  
7 demonstrating good promise as novel heat-resistant alloys. They also investigated the addition of Cu in the  
8 Al-Si-Ni-Fe alloy [205]. The yield compression stress is 355 MPa at 200 °C due to the refined  
9 microstructure as a result of Si, Al<sub>5</sub>Fe(Ni,Cu) and Al<sub>3</sub>(Ni,Cu) phase precipitation (**Figure 14(b)**). Croteau  
10 *et al.*[206] designed an Sc-free, age-hardenable aluminum alloy Al-Mg-Zr specifically tailored for L-PBF.  
11 The developed Al-Mg-Zr alloy was successfully fabricated via L-PBF without cracks and had an excellent  
12 combination of high yield strength (~354 MPa) and high tensile ductility (~20%) after aging due to solid  
13 strengthening by Mg in the matrix and the Hall-Petch effect of both sub-micrometer and nano-sized Al<sub>3</sub>Zr  
14 precipitates. Li *et al.*[123] studied the effect of Zr content on the printability, microstructure evolution, and  
15 mechanical properties of Al-Cu-Mg-Mn alloy. With the increase of Zr content from 0 to 4 wt%, the grain  
16 size decreased and (**Figure 14(c)**) the yield strength and ultimate tensile stress increased. Nie *et al.* [212]  
17 also studied the effect of Zr content on the Al-Cu-Mg-Mn alloy and found that with the increase of Zr  
18 content, the grain size decreased first and then tended to be stable (**Figure 14(c)**) while the yield strength  
19 and ultimate tensile stress increased first and then decreased.



**Figure 14.** Microstructure refining by adding elements that react with the alloy matrix to form nucleating agents. (a) BSE images showing cross section morphologies of (i) AlScZr and (iii) AlErZr after laser remelting; microstructures of laser remelted regions of (iii) AlScZr and (iv) AlErZr; microstructures of casting regions of (v) AlScZr and (vi) AlErZr. EBSD images of grain morphology on cross sections of laser remelted (vii) AlScZr and (viii) AlErZr alloys [203]. (b) Microstructure of Al-Si-Ni-Fe alloy (i, ii) aged for 5 hours at 250°C and annealed for 3 hours at 495°C, and (iii, iv) aged for 5 hours at 250°C [205]. (c) EBSD maps of Al-4.24Cu-1.97Mg-0.56Mn alloy with different Zr contents, (i) 0 wt.%, (ii) 0.6 wt.%, (iii) 2 wt.%, (iv) 2.5 wt.% [212].

In summary, microstructure-oriented strategies are employed to develop alloys tailored for L-PBF in three main ways:

- (1) Phase transition-guided approach: this involves leveraging the intrinsic heat treatment during L-PBF to induce specific phase transitions. By utilizing thermodynamic and kinetic software to calculate phase diagrams, the desired phase constitutions are achievable.
- (2) Columnar or equiaxed grain structure design: the design of the microstructure focuses on either columnar or equiaxed grain structures, depending on the targeted application. The CET theory is utilized to guide the design by optimizing scanning parameters and controlling the thermal gradient.
- (3) Addition of nucleating agents: nucleating agents are added to the alloy to refine the microstructure. This strategy considers the interaction of the nucleating agents with the matrix and their influence on solid solubility and precipitation behavior of other elements in the alloy.

### 5.3 Machine learning-assisted strategies

In section 5.1 and 5.2, we discussed two primary strategies for alloy design in L-PBF. These strategies rely on a deep understanding of physical mechanisms such as cracking mechanism and phase transitions. In order to accurately design the composition of an alloy with specific properties, validated models or theories are required that can link the composition with defects, microstructure, and properties. Unfortunately, at present, either the cracking mechanism or the phase transition during the complex thermal cycles in L-PBF has not been completely understood. Consequently, a validated correlation between alloy composition and final properties has not been well-established, restricting the alloy development for L-PBF.

In recent years, machine learning (ML), a method to create logistic models from large datasets following specific algorithms, has attracted a lot of attentions in diverse industries. Since ML does not solve complex physical equations based on mechanistic models, the computation is fast and is believed to be capable of making more accurate predictions than human beings [213]. Through learning from the literature data, machine learning enables the prediction of material compositions and properties.

Significant advancements have been achieved in the development of ML-driven metals [88, 214-221]. For instance, Zhang *et al.* [222] successfully employed machine learning to develop four novel Cu-In alloys that possess both high ultimate tensile strength and electrical conductivity (EC). These alloys have the potential to replace costly Cu-Ag alloys currently used in railway wiring. Furthermore, Zhang *et al.* [223] combined feature screening strategy with Bayesian optimization to iteratively design alloys. This approach led to the development of a copper alloy, Cu-1.3Ni-1.4Co-0.56Si-0.03Mg, which exhibits exceptional mechanical and electrical properties. Wang *et al.* [224] proposed a machine learning design system (MLDS) that enables performance-oriented compositional design for complex alloys. The MLDS incorporates machine learning modeling, composition design, and property prediction. Through this system, several high-performance copper alloys were designed, offering an ultimate tensile strength range of 600 ~ 950 MPa and electrical conductivity of 50.0% IACS. To further enhance performance-oriented design, Jiang *et al.* [225] extended the MLDS and successfully designed a new type of aluminum alloy with remarkable ultimate tensile strength (707 ~ 736 MPa) and elongation (7.8% ~ 9.5%). Additionally, Xue *et al.* [226] optimized the composition of Ti-Ni-based shape memory alloy using an active learning strategy and discovered a new alloy, Ti-Ni-Cu-Fe-Pd, with reduced transformation thermal hysteresis (1.84 K).

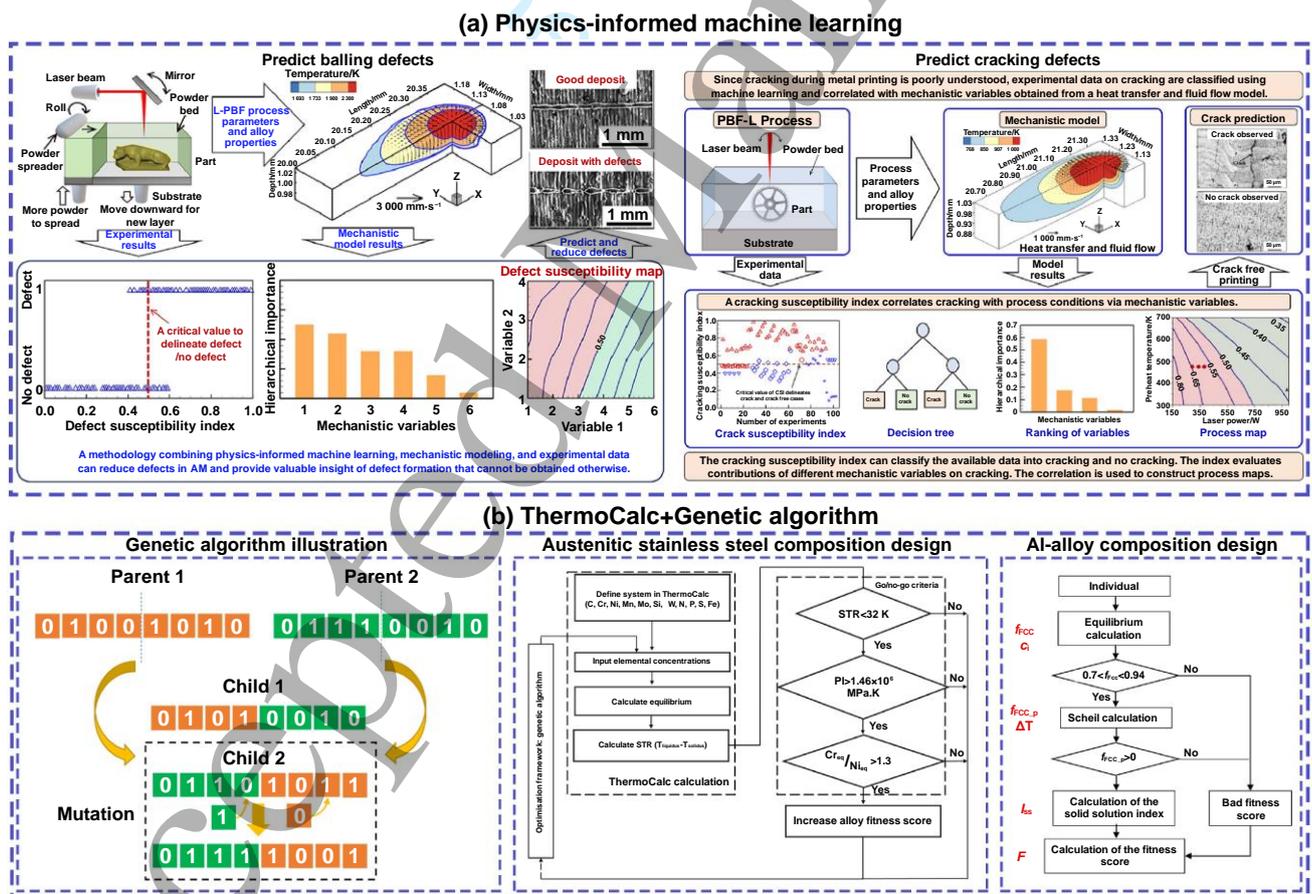
1 Successful applications of machine learning in alloy design can also be found in the development of high  
2 entropy alloys [227], aluminum alloys [228], and Co-based superalloys [214] that undergo conventional  
3 casting or forging processes.  
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7 However, the utilization of machine learning in metal additive manufacturing remains limited. Existing  
8 literature on ML in the AM field primarily concentrates on optimizing processing parameters [229-231],  
9 controlling part geometry [232-234], and detecting defects [234-237]. The incorporation of ML for alloy  
10 design specifically for additive manufacturing is notably scarce in current research.  
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16 Du *et al.* [218] reduced the defects during L-PBF by combining physical-informed machine learning,  
17 mechanistic modeling, and experimental data (**Figure 15(a)**). In order to predict the spheroidizing defect  
18 during L-PBF, parameters such as bulk energy density ( $E$ ), surface tension ( $F$ ), Marangoni number ( $M$ ),  
19 Richardson number ( $R$ ), the aspect ratio of the molten pool ( $\epsilon$ ), and solidification time ( $T$ ) were taken into  
20 account. The spheroidizing sensitivity index was deduced by a genetic algorithm. Two key factors affecting  
21 spheroidization were thus found in this way, i.e., Marangoni number and the solidification time of the  
22 molten pool. They formulated an equation relating the spheroidizing sensitivity index with the  
23 aforementioned six parameters. With this equation, they successfully predicted the spheroidizing defects  
24 with an accuracy of 90%. Mondal *et al.* [221] also applied ML in crack prediction, as shown in **Figure**  
25 **15(a)**. Four parameters were used in this ML model, including cooling rate ( $T$ ), the ratio of the temperature  
26 gradient to the solidification growth rate ( $\epsilon$ ), solidification stress ( $\sigma$ ), and the ratio of the vulnerable and  
27 relaxation times ( $\beta$ ) [221]. The four variables were selected by Pearson correlation analysis. They are  
28 independent of each other and can represent the physical characteristics of the cracks [238, 239]. Ultimately,  
29 a linear correlation was proposed to calculate the crack sensitivity index (CSI).  
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46 Computational alloy design is being widely utilized in various applications. The integration of tools like  
47 Thermo-Calc can significantly enhance the efficiency of alloy development. In cases where direct physical  
48 predictions are impractical, machine learning methods can be employed to forecast properties based on  
49 existing literature data. Multi-objective genetic algorithms are capable of achieving this objective, as shown  
50 in **Figure 15(b)**. The genetic algorithm is an optimization algorithm that emulates biological evolution  
51 processes to iteratively improve solutions to a given problem. The genetic algorithm evolves the  
52 chromosomes through genetic operations such as selection, crossover, and mutation. These operations  
53 simulate the natural selection process, where individuals with better fitness (i.e., solutions that perform well  
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on the given problem) have a higher probability of being selected for reproduction (**Figure 16(b)**- Genetic algorithm illustration). Dreano *et al.* [217] employed qualitative constraints in their study, aiming to control the fraction of face-centered cubic (FCC) phase within the range of 0.7 ~ 0.94 and ensure that the solidification process starts with the formation of FCC phase. They utilized a genetic algorithm that aimed to minimize the solidification temperature range and maximize the strength index. Alloys meeting all the constraints were evaluated based on their fitness scores. In this way, a crack-free aluminum alloy during L-PBF was designed (**Figure 15(b)**-Al-alloy composition design). Similar methods have also been applied in the design of crack-free austenitic stainless steels for L-PBF (**Figure 15(b)**-Austenitic stainless steel composition design). Sabzi *et al.* [88] defined constraints and goals for the genetic algorithm, including the solidification temperature range, performance index (PI), and the ratio of Cr equivalent to Ni equivalent. They optimized the composition of a 316L alloy and validated its printability. The results demonstrated a significant improvement in the formability of crack-free L-PBF 316L alloy compared to 316L alloys with pores and deformation.



**Figure 15.** The application of machine learning in printing crack-free alloys in additive manufacturing. (a) Schematic diagram of machine learning model to avoid balling and cracks [218, 221]. (b) Schematic diagram of multi-objective genetic algorithm combined with thermodynamic calculation for designing aluminum alloys and austenitic stainless steels [88, 217].

To summarize, the use of machine learning in alloy design for additive manufacturing is still in its early stages. This is attributed to the lack of an accurate and statistically representative database, which is essential for the prediction accuracy of machine learning. Furthermore, many mechanistic variables of alloys might not be readily available and thus should be calculated based on known data. For instance, temperature gradient, cooling rate, and stresses, which influence the solidification microstructure and cracking, should be calculated by combining CALPHAD-based computational thermodynamic and finite-element-based thermal models. These mechanistic variables are important inputs in the ML model construction.

## 6. Future outlook and conclusions

Since the emergence of additive manufacturing (AM) in the 1980s, a considerable amount of work on metal additive manufacturing has been published. Despite significant progress in AM, some scientific and technological challenges still need to be addressed. For instance, cracking, anisotropy in building direction, low ductility, and low fatigue resistance constantly occur in additively manufactured alloys, particularly in the LPBF-processed alloys. The previous work has proposed various remedies to tackle these issues during alloy design for L-PBF, as shown in **Figure 16**. However, printable metals with satisfactory mechanical properties represent only a small fraction of commercially available alloys. The following aspects should be emphasized in future research.

- (1) The defect formation mechanism in L-PBF requires clarification. Although our understanding of the L-PBF process has significantly advanced (e.g., RGD theory, Kou's criterion), the root cause of cracking during L-PBF has not been completely understood. Due to the complex metallurgical reactions (laser-powder-gas-bulk interactions) during L-PBF, a practical in-situ crack observation technology that can visualize the crack initiation and propagation during L-PBF is in urgent need.
- (2) Solidification theories tailored for the L-PBF process need to be developed. The layer-by-layer powder deposition in L-PBF facilitates epitaxial grain growth, which is detrimental to the property isotropy in the final component. Considering the L-PBF process features, a solidification theory accounting for fast cooling, steep temperature gradients, and huge internal stress is required. Additionally, tailored thermodynamic packages that consider the phase transitions during thermal cycles in L-PBF are necessary.

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- (3) Further exploration of grain boundary segregation engineering is needed during alloy design for L-PBF. Grain boundary manipulation is crucial for the development of crack-resistant alloys, as cracks primarily initiate from grain boundaries. It is important to note that the sub-grain boundary (cell/dendrite) segregation also plays a significant role in crack formation. The enrichment of solute atoms can lead to precipitations, which reduce grain boundary energy and hinder boundary mobility through pinning mechanisms. This pinning effect promotes the formation of equiaxed grains but also introduces stress concentration. Care must be taken when assessing the influence of precipitates on crack sensitivity due to their embrittling nature.
- (4) Further research is needed to apply machine learning in alloy design for additive manufacturing, accompanied by the high-quality database. The application of machine learning in alloy design for additive manufacturing is still in its nascent stages. This is primarily due to the limited volume of data available for ML in the additive manufacturing field. Additionally, data consistency in AM is unsatisfactory. For instance, the crack density of a sample can vary significantly when printed with different machines or using different batches of powders. Furthermore, more efforts should be made towards developing physics-informed machine learning models that aid in understanding the cracking and strengthening mechanisms of LPBF-processed alloys.
- (5) High-throughput L-PBF equipment or methods are in urgent need, taking into account the time-consuming and costly powder making and conventional L-PBF processes. In-situ alloying presents a high-throughput methodology to design novel alloys for additive manufacturing. By combining multiple elemental powders during printing, engineers can design and test various alloy compositions to achieve desired material properties, reducing the need for pre-alloyed powders that may have limited availability or high costs. By blending elemental powders during printing, the desired alloy composition can be achieved precisely where and when it is needed, providing greater flexibility in material selection. This flexibility in composition enables the rapid exploration of new alloy systems. For instance, blends of titanium and Cr powders have been manufactured using directed energy deposition (DED), and a desirable combination of strength and ductility was obtained [240]. Grigoriev *et al.* mixed Ti, Al, and Nb elemental powders in a turbula mixer and processed the mixture with L-PBF to produce  $Ti_2AlNb$  parts [241]. However, incompletely melted and unevenly distributed Nb particles were observed in the printed specimen due to the high melting point of Nb (2 469 °C). Balancing the melting of elements with high melting points while avoiding the evaporation of volatile

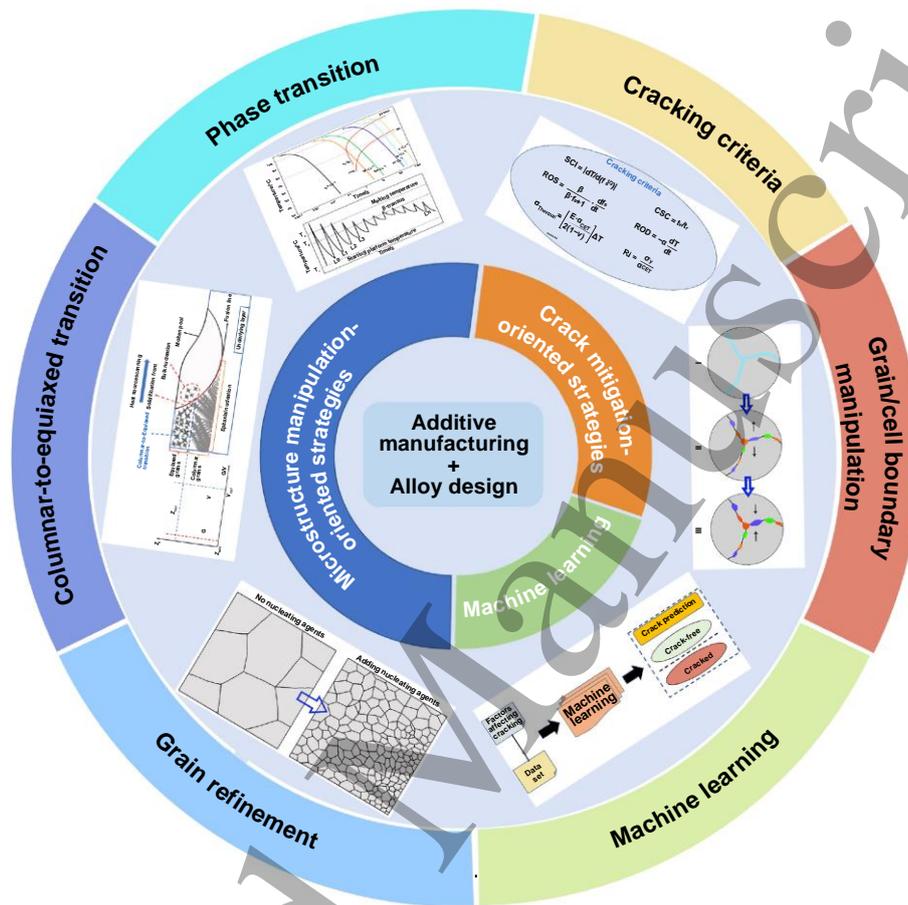
elements during parameter optimization remains a challenge in in-situ powder mixing for L-PBF.

This review assessed the printability of four commonly used alloy systems (Fe-, Ni-, Al-, and Ti-based alloys) mainly in laser powder bed fusion additive manufacturing and compared with their weldability. The application of weldability criteria in assessing printability for each alloy was also evaluated. Various pathways for AM alloy design were reviewed and analyzed. The following conclusions were drawn:

- (1) The presence of martensite in the as-printed steel microstructure increases crack susceptibility. The Schaeffler diagram can be used to estimate the printability of steels based on their chemistry.
- (2) The weldability of nickel-based superalloys does not always align with their printability. The weldability assessment chart, which considers the effects of Ti, Al, Cr, and Co, cannot directly categorize the printability of nickel-based superalloys during L-PBF.
- (3) A Pearson correlation map was constructed to identify key factors contributing to cracks in aluminum alloys processed via L-PBF. The map revealed that increasing silicon content and decreasing magnesium content can enhance crack resistance in aluminum alloys during L-PBF.
- (4) Titanium alloys generally exhibit less severe cracking during L-PBF. Efforts have been focused on promoting the columnar-to-equiaxed transition during the AM of titanium alloys.
- (5) Alloy design strategies for L-PBF can be categorized into three types: 1) crack mitigation-oriented strategies that aim to mitigate cracks through in-depth analysis of the crack formation mechanism, 2) microstructure refining-oriented strategies that focus on synergistically improving crack resistance and mechanical properties by adding alloying elements or nucleation particles, and 3) machine learning-assisted strategies based on AM databases and data analysis.
- (6) To accelerate alloy design for L-PBF, there is a need to enhance fundamental understanding of defect formation, building direction anisotropy, low ductility, and low fatigue resistance in LPBF-processed alloys. Thermo-Calc calculations can be utilized to address complex phase transitions during AM by considering features such as extremely fast cooling and cyclic heat treatment. Grain boundary should be carefully manipulated considering the strengthening and embrittling effects of precipitates. In-situ alloying provides a high-throughput method for alloy design, but challenges related to composition homogeneity and melting point variation among elements need to be resolved.

In summary, the complex metallurgical reactions and phase transitions during L-PBF pose challenges to

the quality of printed parts. However, they also open avenues for revolutionary alloy design, offering the potential for unprecedented metals with tailored properties in the near future. By addressing the L-PBF processability challenges and exploring innovative alloy design strategies, this review contributes to a better understanding of alloy design for laser powder bed fusion additive manufacturing and guides future research.



**Figure 16.** An overview of various alloy design strategies for laser powder bed fusion.

## 7. Acknowledgments

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